



CENTRE NATIONAL D'ÉTUDES SPATIALES

IASI-NG : CNES phase 0 study presentation

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***2nd IASI conference
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- 1- IASI-NG performance objective and compliance with the EUMETSAT POST-EPS IRS Mission Requirement Doc**
- 2- IASI-NG CNES phase 0 technical synthesis**
- 3- IASI-NG CNES phase 0 expected performances and budgets**
- 4- IASI lessons learned from technical point of view**
- 5- IASI-NG activities in 2010**

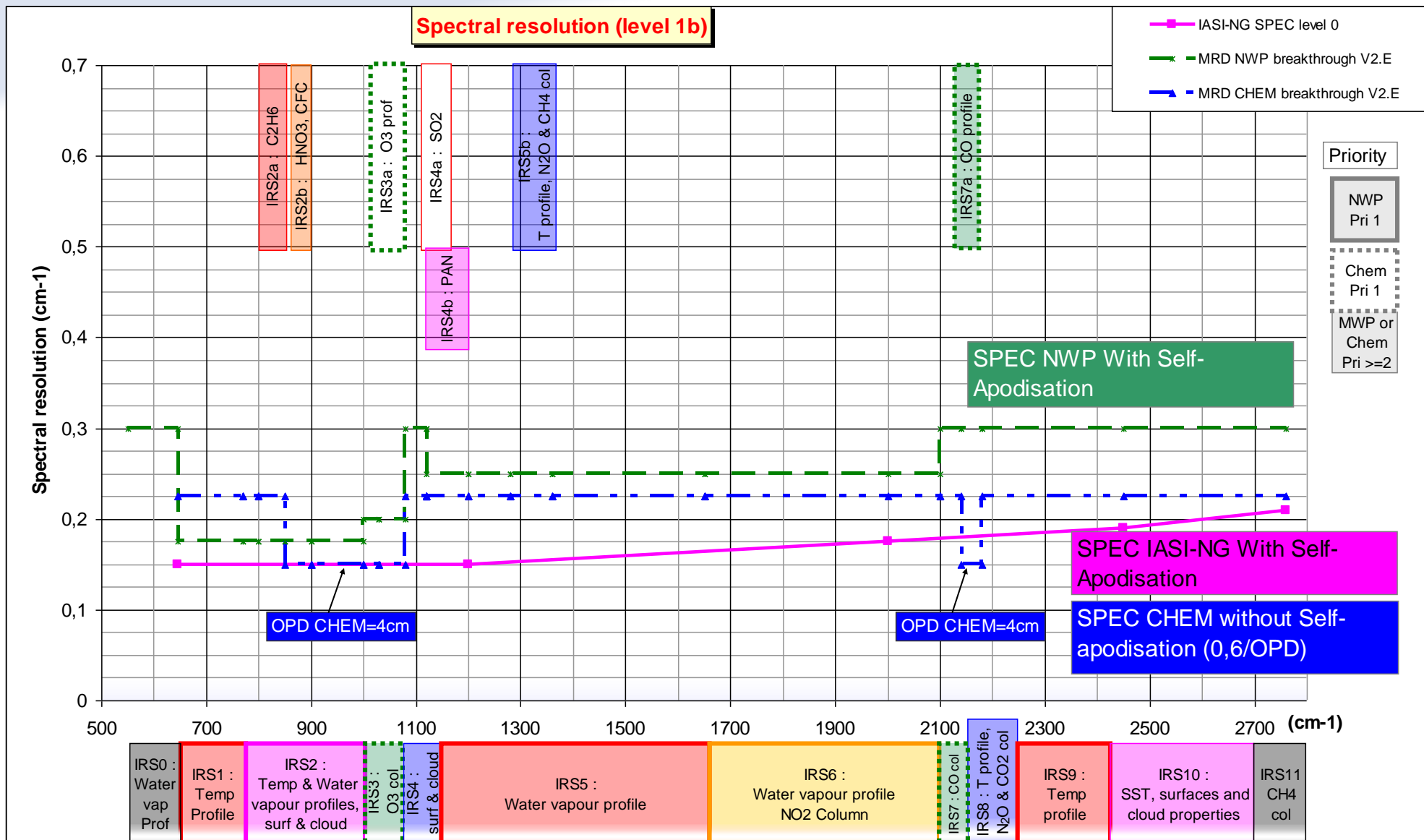
- Improve the IASI demonstrated performances :
 - ♦ Spectral Resolution by factor 2
 - ♦ Radiometric noise by factor 2

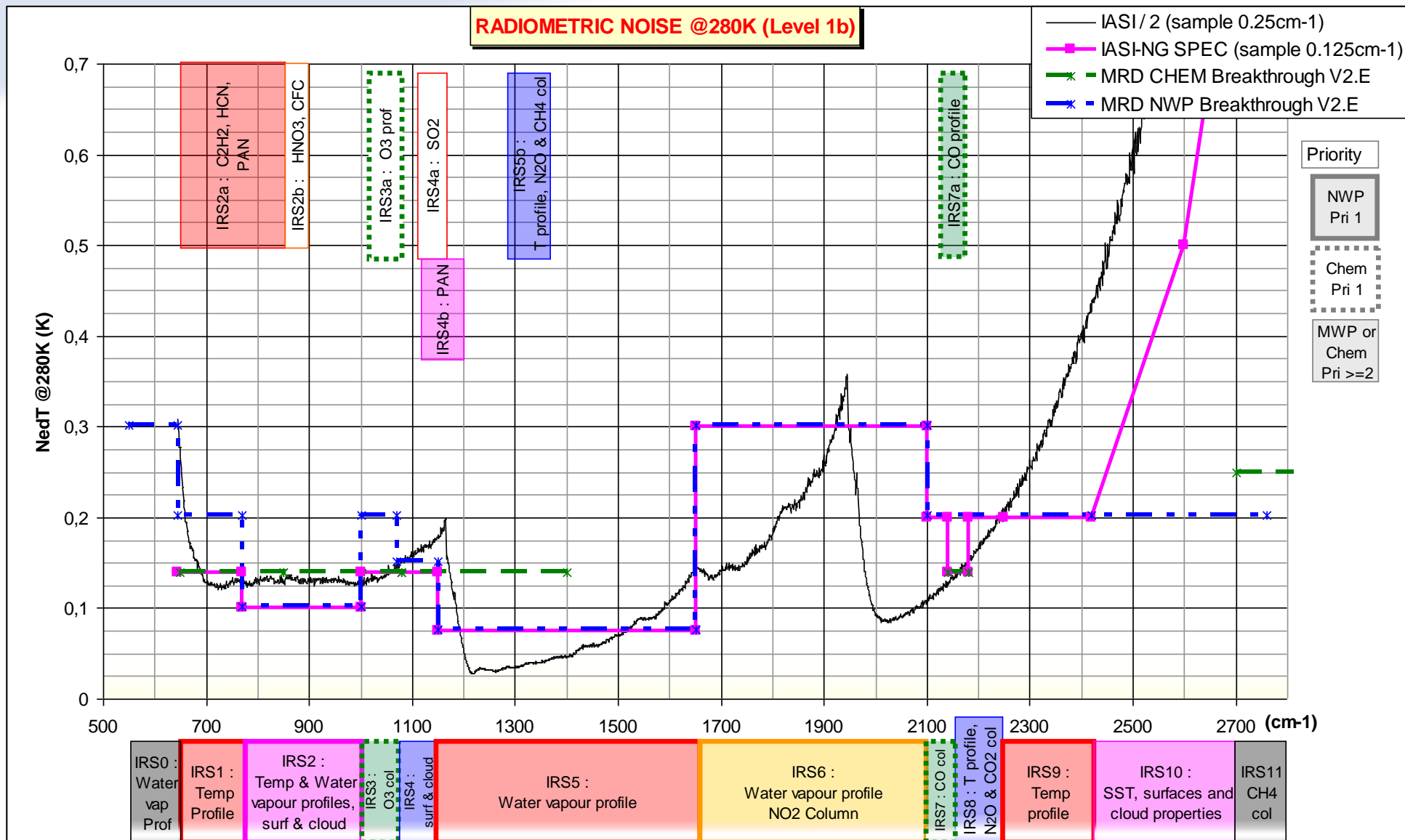
==> comparable to Post-EPS MRD 2.E **breakthrough** data
- Keep the same spatial resolution as IASI (25km in average on ground at nadir) and the same IASI pixel size (12km)
- Compliance of the IASI-NG specification to Post-EPS IRS requirements presented to PMET end of 2009.

Main non compliances with MRD 2.E:

 - ♦ Viewing angle and coverage (will be the IASI one for IASI-NG)
 - ♦ Pointing knowledge (will be just a bit better than the IASI one)
 - ♦ Radiometric noise below 645cm^{-1} and above 2400cm^{-1}

Accepted by PMET : MRD to be updated





Different trade off's done only for dynamic FTS

Radiometry

- Entrance pupil diameter Increase (\rightarrow flux):
 - IASI-NG = 120mm versus IASI = 80 mm
- Instrument Field of view Increase (\rightarrow integration time):
 - IASI-NG = 75*75km (9 pixels) versus IASI = 50*50km (4 pixels) \rightarrow acquisition duration for each interferogram = 450ms versus 150ms for IASI
- Detectors temperature reduction (active cooling)
 - IASI-NG T detector < 65K versus IASI = 92K

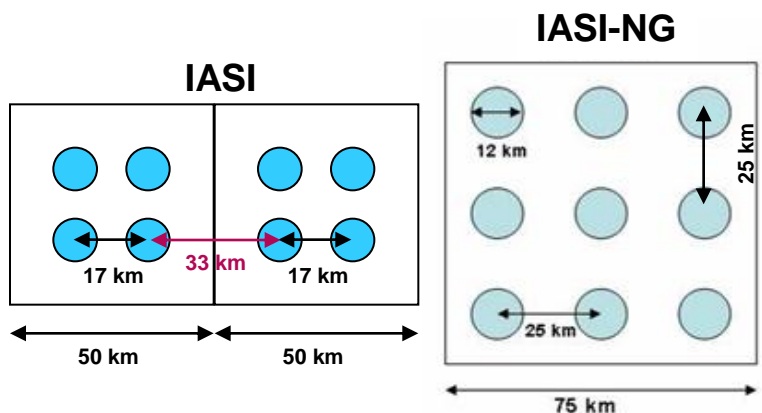
Spectral resolution

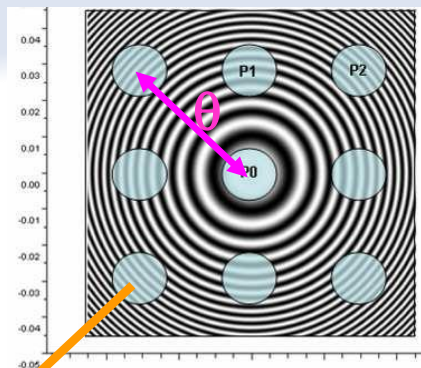
Optical Path difference increase by factor 2:

- single sided interferometer
 - one mobile cube having the same IASI stroke
 - TWO mobile cubes having the IASI stroke / 2
- double sided interferometer
 - one mobile cubes having the IASI stroke x 2
 - TWO mobile cubes having the same IASI stroke

BUT: «self-apodisation»

*For the spectral resolution, both the Optical Path Difference **AND** the self-apodisation must be improved.*





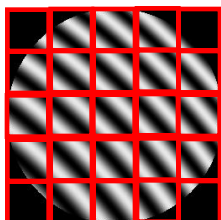
IASI-NG: Optical signal in the detectors plane for $\nu=2760\text{cm}^{-1}$ and $\text{OPD}_{\text{max}}=4\text{cm}$

The use of IASI like pixel acquisition concept is not possible. Two options has been studied.

Split the sounder pixel into many smaller pixels
→ Matrix detectors

Suppress/mitigate the self-apodisation effect
→ self-apodisation compensation

Sounder pixel



- For each of the 9 sounding pixel, acquisition of sub-pixels interferograms (typically 5x5 sub-pixels), then combination of the interferograms by resampling at “constant” OPD + filtering to generate one sounding pixel interferogram
- Final Self-apodisation for the corner pixels = IASI one

Huge Data Processing (1Gbit/s of data at the matrix output)

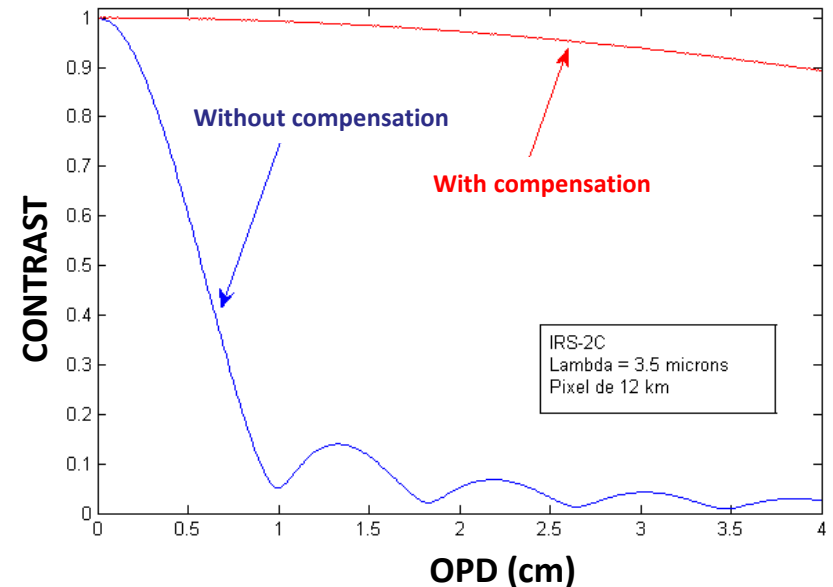
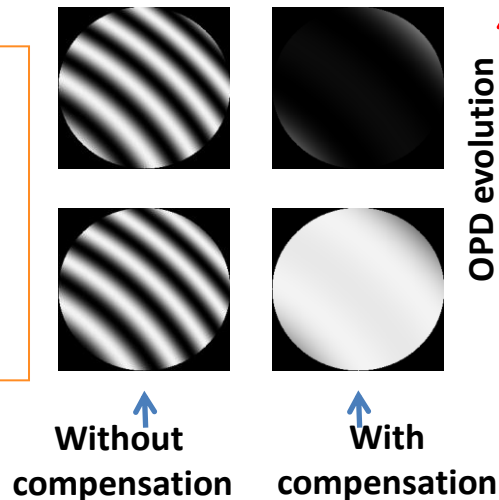
Need a very GOOD focal plane image quality → complex focal plane

- Introduce, in the interferometer, a specific mechanism that works in synchronization with the Corner Cube Mechanism and that corrects, for each sounding pixel (but the center one), the Optical Path Difference by : $\delta_o (1 - \cos \theta)$, where δ_o is the OPD for a zero field → **ALL the pixels should have a similar behavior than the central pixel**

IASI-NG CNES Phase 0
Reference option

Optical simulation with and without active optical field compensation

Evolution of a corner pixel optical signal for monochromatic ray at $4\mu\text{m}$, for OPD variation of $1\mu\text{m}$, (between max and min signal ($\lambda/4$)) at OPD of 4cm



Advantages

Direct acquisition of one interferogram by sounder pixel using monoelement pixel detectors (IASI like) or by matrix detectors with hardware binning of the elementary pixels

Data processing IASI like (excepted for the single sided processing)

IASI like cold optical focal plane

Reduction of the different defaults sensitivity with regard to IASI (CCM, microvib's)

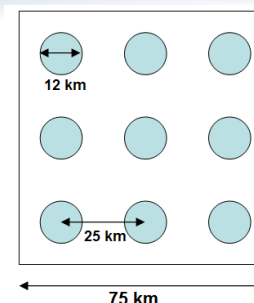
Drawbacks

New Genzel concept

New mechanism

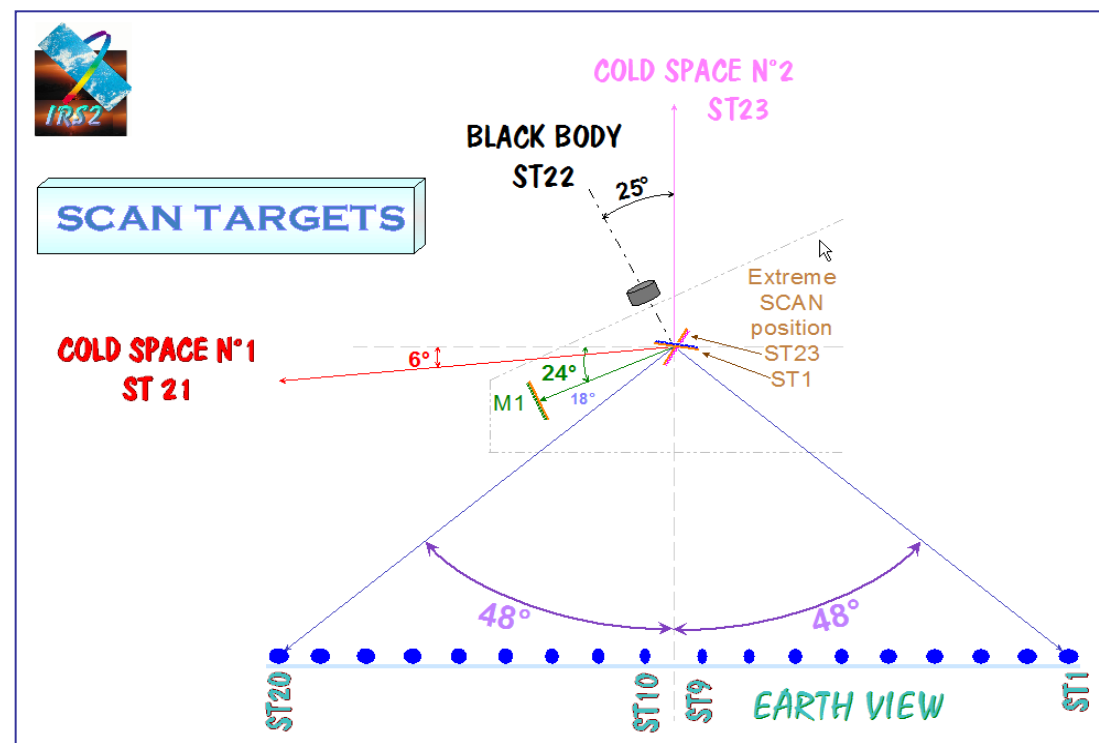
Complexification of the Laser Reference accommodation

- ◆ Ground Pixel diameter of 12km (=IASI)
- ◆ Ground sampling of 25km (both axis)
- ◆ Number of sounder pixels per acquisition = 9 (IASI=4)
- ◆ Number of earth view per line = 20 (30 IASI)
- ◆ Interferogram acquisition duration = 450ms (IASI=150ms)



- ◆ Inlet PUPIL = 120mm (IASI=80mm)

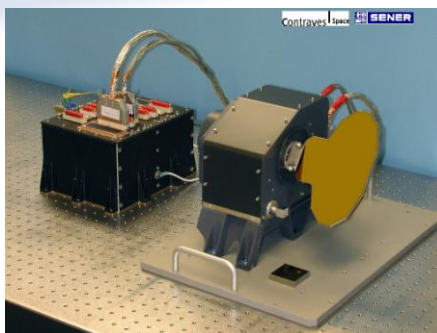
- ◆ Focal plane :
 - 4 bands (IASI=3)
 - 9 sounder pixels per band (IASI=4)
 - PV detectors for all bands (IASI PC for B1; PV for B2/B3)
 - detectors cooled at 65K with one active cooler (LPTC) (IASI=92K passive)
 - IASI like cold optic concept



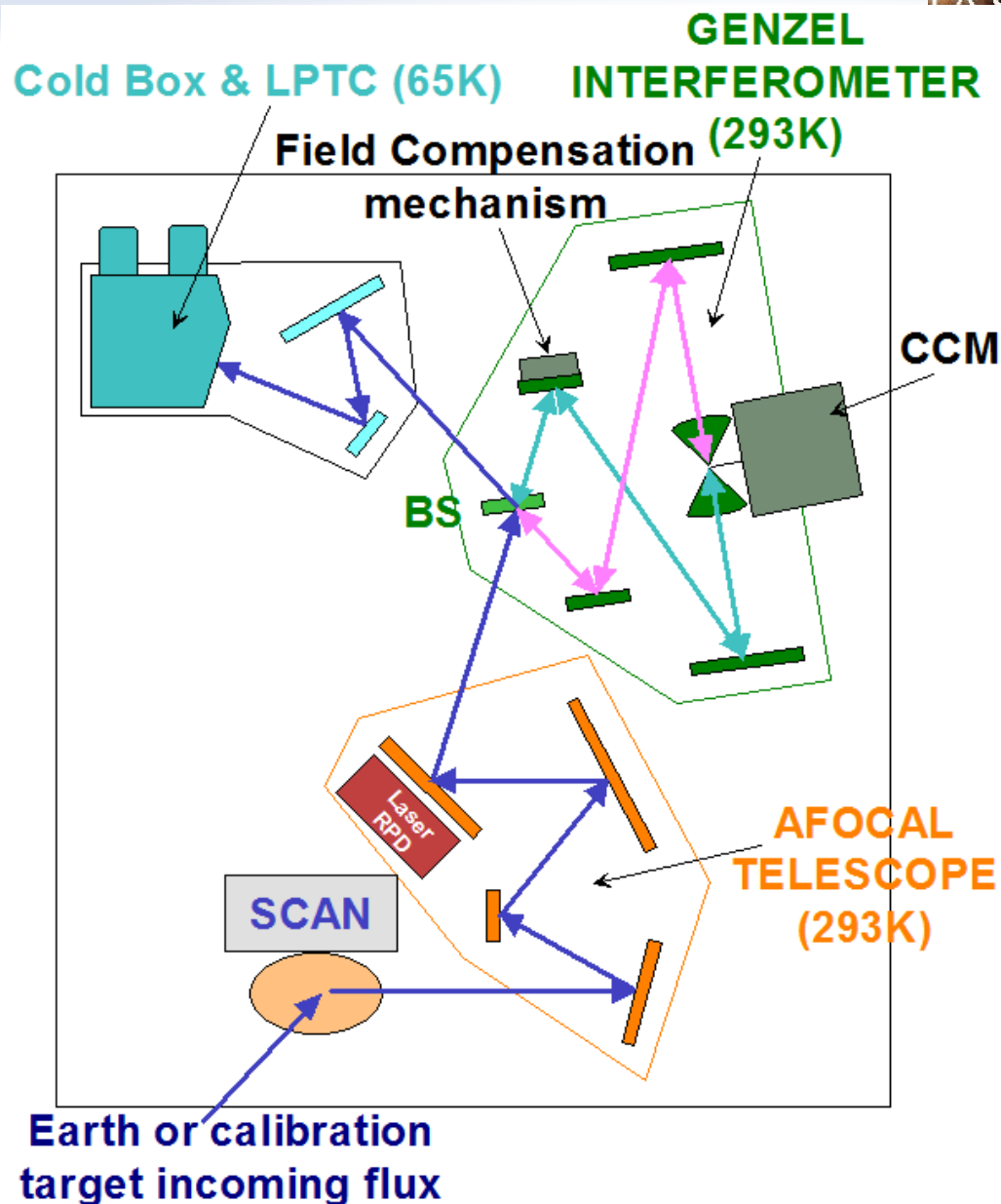
cnes IASI-NG : Baseline characteristics (CNES phase 0)



- ♦ IASI like SCAN with slight increase of the motorization

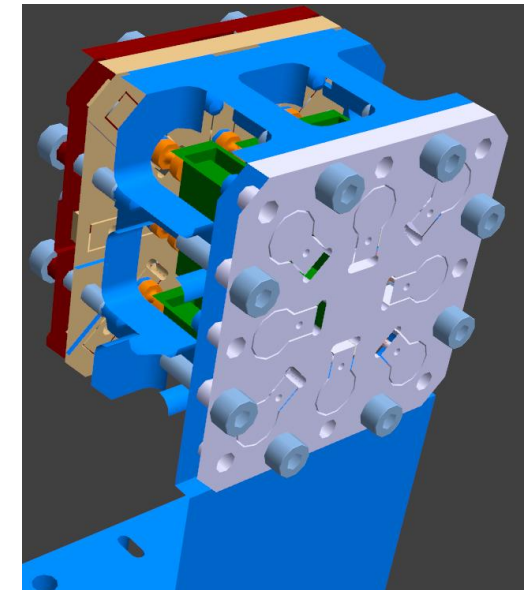
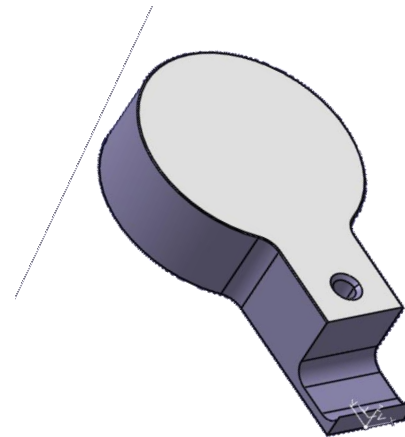
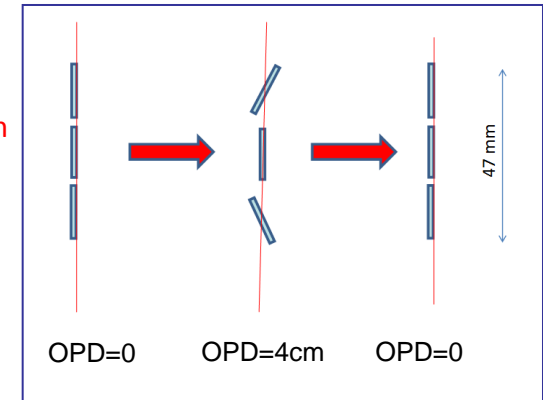
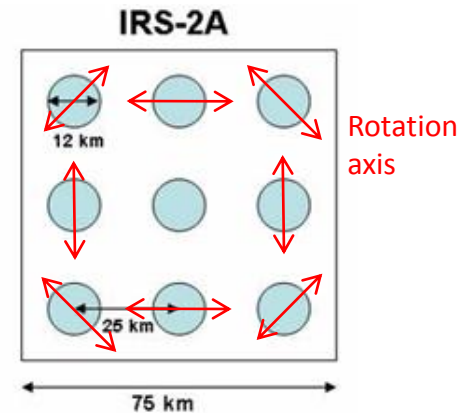


- ♦ One afocal telescope
- ♦ One GENZEL interferometer
 - 1 ZnSe Beam Splitter (80mm typical)
 - Single sided configuration (to be justified more deeply)
 - 2 Corner Cubes are linked together and are activated by a single mechanism (CCM) with a stroke of 11mm (IASI=20mm for one CC)
 - 1 field compensation mechanism
 - One laser beam for OPD sampling for each pixel



♦ **Genzel interferometer :**
Mechanism for field compensation

- 9 individual independent flat mirrors. 8 mirrors are mobile. The central one is not moving.
- Each individual mirror is activated in rotation by one independent piezzo in coordination with the Corner Cube Mechanism (the mirror on the center is not moving) :
 - $OPD=0 \rightarrow$ No rotation; all the mirrors are in the same plane
 - $OPD \text{ max} \rightarrow$ maximum rotation of typically 1.5mrad
- A breadboard is under development at CNES



♦ **Data processing :**

- *Earth precalibrated spectra are downloaded + IR image*
- *DPS with high heritage of IASI for the algorithms excepted for the single sided interferogram processing*
- *LEON's processor + FFT coprocessors (FFT of 65536 samples is done in 1ms)*
- *Spacewire bus*

♦ **IR integrated Imager :**

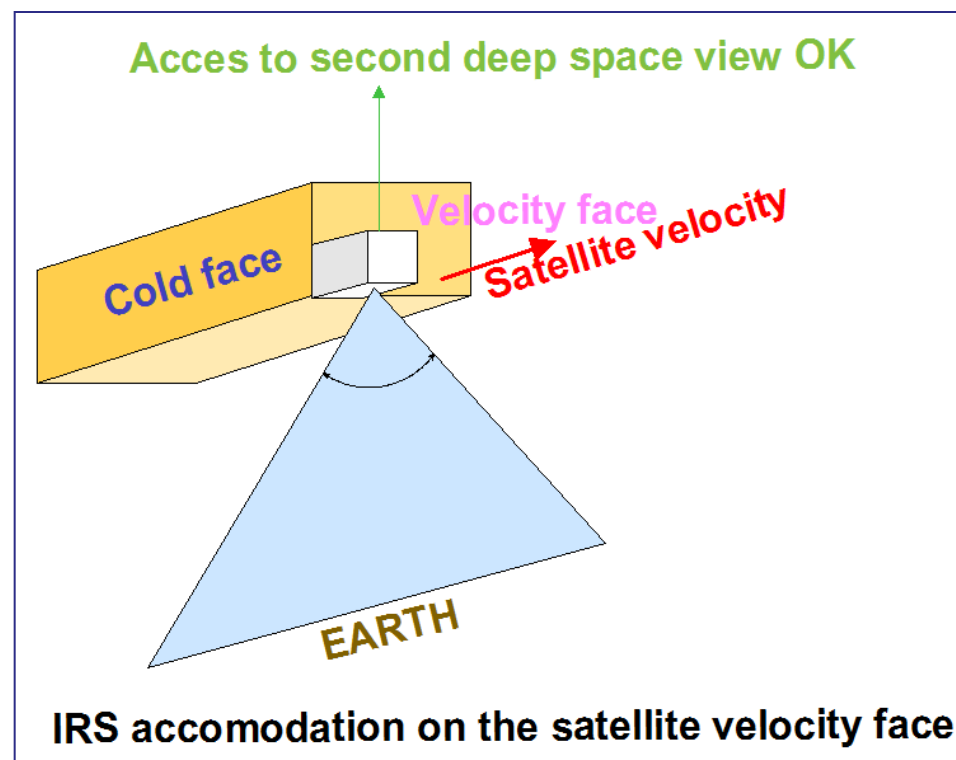
- *IASI like with increased field of view*

♦ **Instrument Management :**

- *Classical with 1553 bus*

♦ **Mechanical & thermal:**

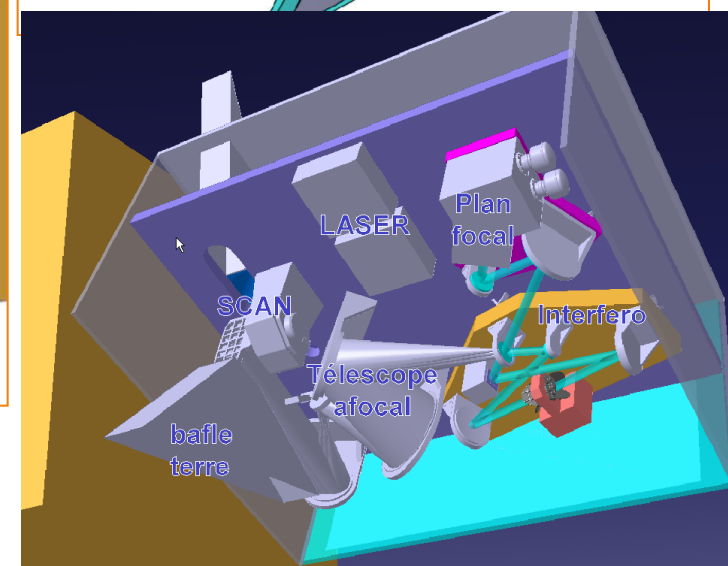
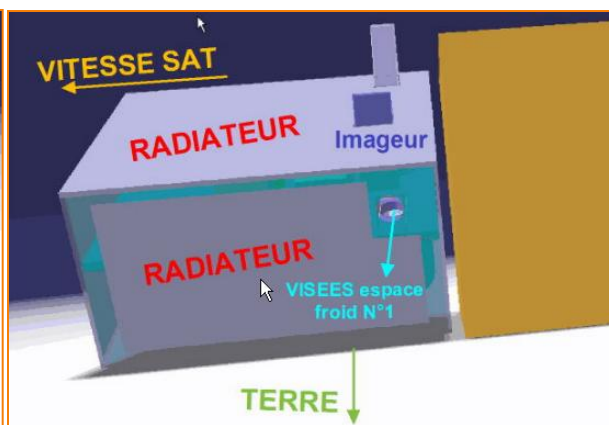
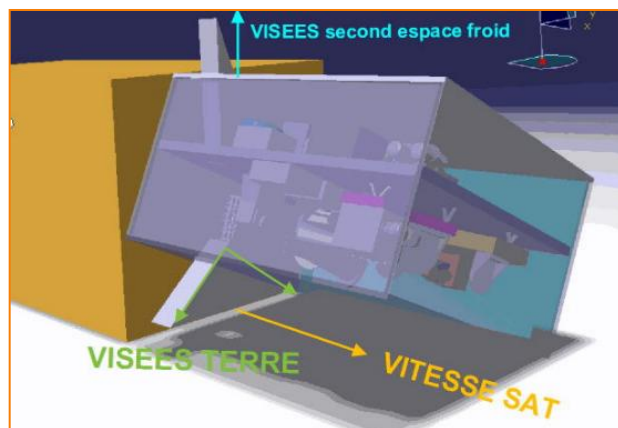
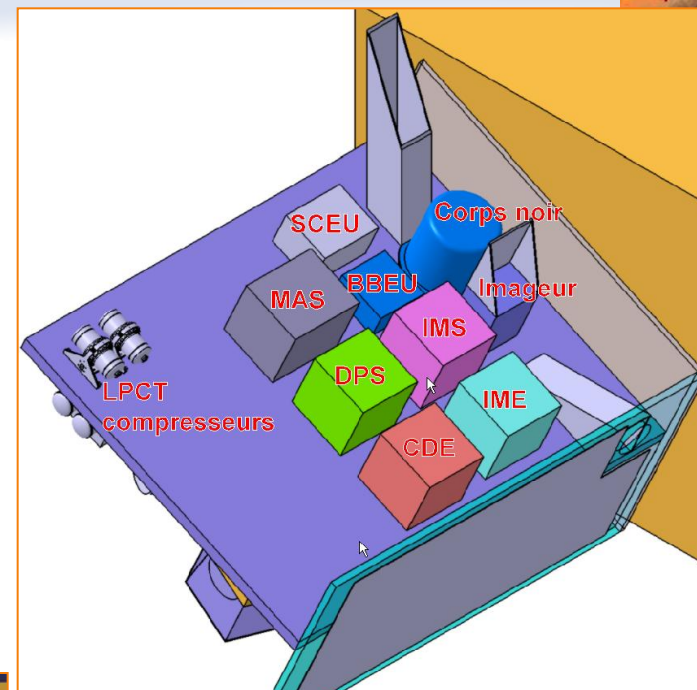
- *IASI-NG accommodation on the satellite velocity face with access to a second deep space calibration view*





♦ **Mechanical & thermal:**

- Interferometer equipments are mounted on a stiff CFRP optical bench
- ALL the others panels are aluminum honeycomb panels
- ALL equipments, panels and baffles are controlled at 293K (excepted the detectors at 65K)
- No Launch Locking and microvibrations Filtering Device



♦ **MASS**

- ♦ **IASI-NG : 260Kg (best estimate without margin)**
- ♦ **IASI = 235Kg**

♦ **VOLUME**

- ♦ **IASI-NG : 1700mm x 1400mm x 800mm in a single box (1,9m³)**
 - Optimization can be envisaged
- ♦ **IASI : one sensor module + deported IMS/DPS inside the satellite ==> total = 1,7m³**

■ **POWER (EOL):**

- ♦ **IASI-NG : 370W (best estimate without margin) (active cooling main contributor)**
- ♦ **IASI = 240W**

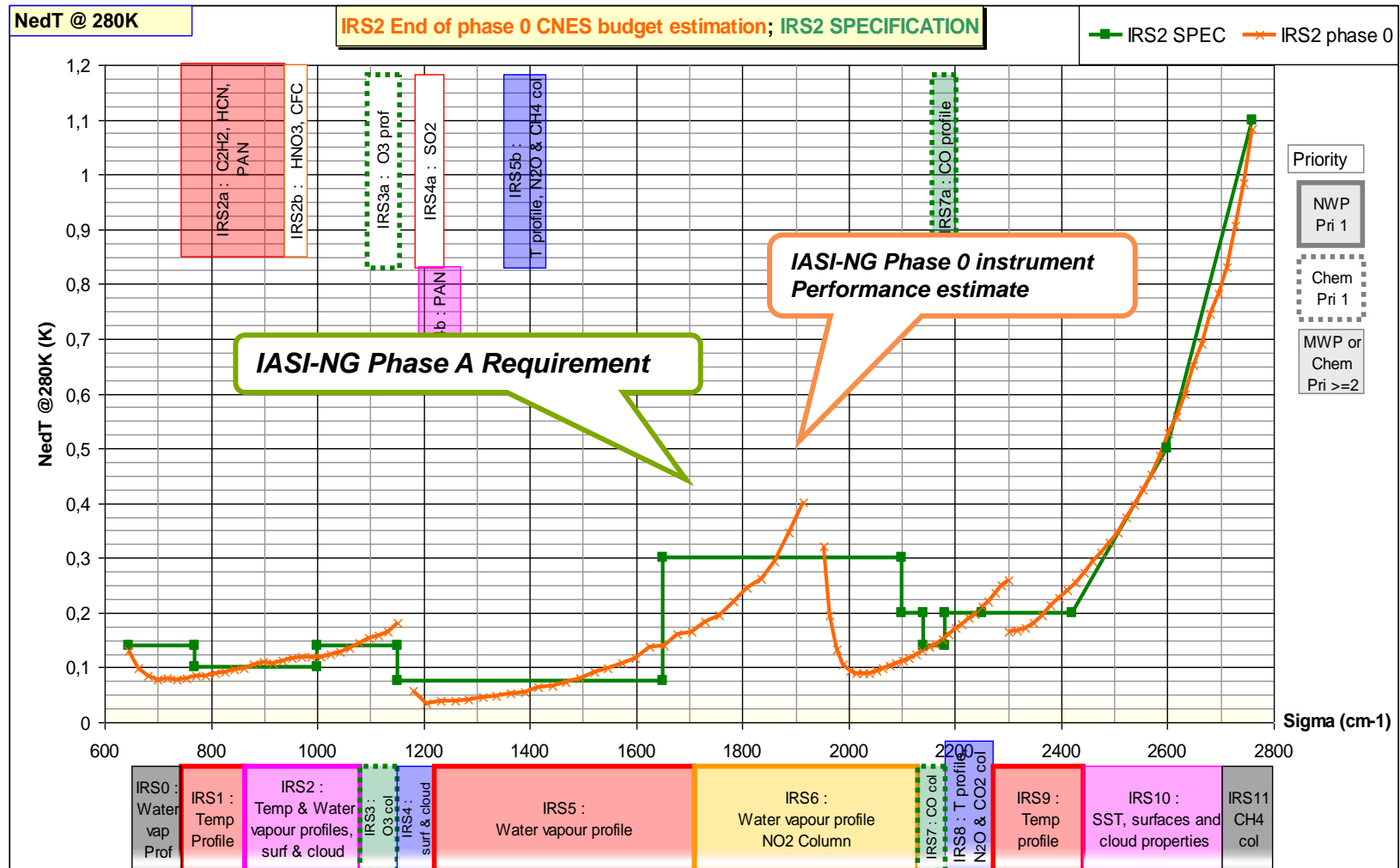
■ **TM DATA RATE:**

- ♦ **IASI-NG : 3.1Mbit/s**
- ♦ **IASI : 1.5Mbit/s**

cnes IASI-NG : Baseline characteristics (CNES phase 0)



- ♦ Spectral resolution at level 1b = 0.15cm^{-1} almost constant all over the spectral bandwidth. = 0.25cm^{-1} at level 1c (after numerical apodisation)
- ♦ Radiometric noise (level 0 for samples of 0.125cm^{-1}):

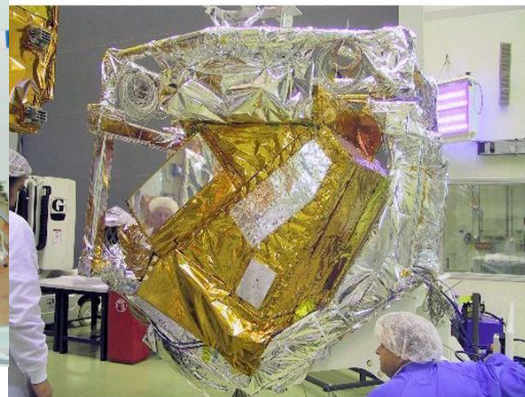


■ **Engineering & AIT constraints improvements / IASI**

- ♦ **Thanks to a very stiff interferometer :**
 - The instrument performances should not be sensitive to gravity
 - The instrument performances should be much less sensitive to microvibrations
- ♦ **The corner cube compensation mechanism (<10Hz) should be avoided (one main source of microvibration for IASI)**
- ♦ **Thanks to an active cryocooler and a sealed cold box/focal plane:**
 - the instrument should be able to cool the detectors by itself at ambient cleanroom condition (Possibility to do representative EMC tests at ambient, ...)
 - No need of the quite complex IASI test jig and cryopannel for the satellite TV tests
 - The ice contamination of the cold optics should be minimized
 - Can fly on any sun-synchronous orbit <10h30 and >13h30 with minor modification on the warm thermal radiators

IASI configuration for TV test !!

IASI flight configuration



■ Engineering & AIT constraints improvements / IASI

- ◆ Should avoid Locking and Filtering device → no storage and transportation constraints
- ◆ The CCM locking/unlocking mechanism should be simplified
- ◆ The instrument shall be able to be fully functional with the satellite in vertical position and for one horizontal position
- ◆ No deported equipments (IMS/DPS for IASI) inside the satellite
- ◆ Easy accessibility to all of the equipments including the cold focal plane should be possible

■ Operational improvements / IASI

- ◆ The unavailability period due to ice decontamination should be significantly reduced
- ◆ Following an anomaly, the recovery time should be significantly reduced

- ♦ **A competitive industrial IASI-NG instrument phase A has started with ASTRIUM and THALES beginning of 2010.**
- ♦ **Both industries will evaluate the CNES phase 0 concept AND some others ones. A concept selection will be done before mid 2010 for more deep analysis studies.**
- ♦ **The end of this IASI-NG phase A is expected for beginning of 2011.**
- ♦ **Discussion start with EUMETSAT about IASI-NG ground segment**