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Sheet: 1

Of:

49

IASI QUARTERLY PERFORMANCE REPORT FROM 2020/03/01 TO 2020/05/31

Metop-B

BY IASI TEC (TECHNICAL EXPERTISE CENTER)

FOR IASI PFM-R ON METOP B





Doc n°:

IA-RP-2000-4510-CNE

Issue:

1.0

2

Date: Sheet: 01/06/2020 Of:

49

ANALYSE DOCUMENTAIRE

Metop-B

Bordereau d'indexation

Classe (Confidential	lité) : NC	Code Consultation :			
Mots clés d'auteur :	IASI TEC quarterly synthes	is report			
OBJET:	IA	SI TEC	periodic report		
TITRE : IASI quarterly performance report					
Auteur(s) : Mathilde Faillot et Laura Le Barbier (CNES) avec le support de Catherine Sorignet et J.C. Calvel (AKKA Technologies, marché AC-SIS 171475)					
RESUME: Quarterly report issued by the IASI TEC team to show trends and layout from the "long term synthesis" TEC function for flags and observables quality indicators					
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Issue: 1.0

Date: 01/06/2020

Sheet: 3 Of: 49

DIFFUSION

Metop-B

On CNES web site : https://iasi.cnes.fr Instrument characteristics / In-orbit performances monitoring

DOCUMENT MODEL CHANGE RECORD

Version	Date	Paragraphs	Description	
1.0	05/03/11		Creation of the model	
2.0	01/01/15	4.3.2.5	Cube corner Speed Quality (CSQ) monitoring	
		4.8	IASI-B inter-calibration with IASI-A, CrIS and AIRS	
2.1	01/03/16	4.6.2	Spectral calibration will be synthesized only once per year (in REVEX report)	
		4.6.3	Monitoring of the Ghost evolution is stopped (there is no more ghost since CD stop)	
		4.9	IIS noise and IIS radiometric calibration will be computed in one monthly external calibration in three and monitored only once per year (in REVEX report)	
2.2	01/12/2017		OpenOffice Writer → LibreOffice Writer 5	
3.0	01/12/2020	4.8 5.2	IASI-C inter-calibration with other instruments will be detailled in a new specific document. (old paragraphs §4.8 : IASI-C Inter-calibration and §5.2 : SIC software evolution are removed)	

DOCUMENT CHANGE RECORD

Version	Date	Paragraphs	Description
1.0	01/06/2020		Creation of the document





Issue: 1.0

Date: 01/06/2020

Sheet: 4 Of: 49

Table of contents

Metop-B

1	INTRODUCTION	9
2	RELATED DOCUMENTS	9
	2.1 APPLICABLE DOCUMENTS	g
	2.2 REFERENCE DOCUMENTS	g
3	SIGNIFICANT EVENTS	10
	3.1 EXTERNAL CALIBRATION	
	3.2 ON BOARD CONFIGURATION	
	3.3 GROUND CONFIGURATIONS UPDATES FOR LEVEL 1 PROCESSING	
	3.4 DATA BASES UPDATE FOR THE USERS	
	3.5 ON GROUND HW/SW EVOLUTION	
	3.6 DECONTAMINATION	
	3.7 INSTRUMENT	
	3.7.1 External events	
	3.7.2 Operation leading to mission outage	13
	3.7.3 Anomaly leading to mission outage	
4	PERFORMANCE MONITORING	14
	4.1 PERFORMANCE MONITORING	14
	4.2 PERFORMANCE SYNOPSIS	
	4.3 LEVEL 0 DATA QUALITY (L0)	16
	4.3.1 Overall quality	
	4.3.2 Main flag and quality indicator parameters	
	4.3.3 Second level flags and quality indicators	
	4.4 LEVEL 1 DATA QUALITY (L1)	
	4.4.1 Overall quality	27
	4.4.2 Main flag and quality indicator parameters	
	4.4.3 Conclusion	
	4.5 SOUNDER RADIOMETRIC PERFORMANCES	
	4.5.1 Radiometric Noise	
	4.5.3 Delay of Detection Chains	
	4.5.4 Optical Transmission	
	4.5.5 Interferometric Contrast	
	4.5.6 Interferogram Baseline	
	4.5.8 Conclusion.	
	4.6 SOUNDER SPECTRAL PERFORMANCES	
	4.6.1 Monitoring of the ISRF inputs	
	4.6.2 Spectral calibration assessment	
	4.6.3 Conclusion.	
	4.7 GEOMETRIC PERFORMANCES	
	4.7.1 Sounder / IIS co-registration monitoring	
	4.7.2 IIS / AV FIRR CO-legistration	





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 5 Of: 49

4.8 IIS RADIUMETRIC PERFORMANCES	
4.8.1 IIS Radiometric Noise Monitoring	
5 IASI TEC SOFTWARE AND INTERFACES	
5.1 IASI TEC EVOLUTION	
5.2 EUMETCAST INTERFACE	48
5.3 FTP INTERFACE	48
6 CONCLUSION AND OPERATIONS FORESEEN	49
6.1 SUMMARY	49
6.2 SHORT-TERM EVENTS	49
6.3 OPERATIONS FORESEEN	
Figures and tables	
Table 1: External Calibration TEC Requests	
Table 2: DPS and MAS configuration TEC Requests	
Table 3: DPS and MAS previous configuration	
Table 4: IASI L1 Auxiliary File Configuration on the Operational EPS Ground Segment	
Table 5: IASI L1 auxiliary file previous configuration	
Table 6: IASI Data Bases for the users	
Table 7: previous IASI Data Bases	
Table 8: IASI L1 PPF Configuration on the Operational EPS Ground Segment	
Table 9: Previous IASI L1 PPF	
Table 10: Decontamination TEC Requests	12
Table 11: Previous decontamination	
Table 12: Overview of METOP manoeuvres in the reporting period	12
Table 13: PL-SOL	13
Table 14: Scheduled interruptions.	13
Table 15: Major anomalies	13
Table 16: Minor anomalies	13
Table 17: Functional status legend	14
Table 18: IASI product components functional status	15
Figure 1 : IASI L0 data quality orbit average (per pixel and CCD)	16
Figure 2 : IASI L0 data quality spatial distribution (per pixel)	17
Figure 3: Temporal evolution of spikes anomaly ratio in % for all pixels (orbit average)	18
Figure 4: Geographical distribution of spikes occurrences in % for band 3 and all pixels	19
Figure 5: Temporal evolution of NZPD determination anomaly ratio in % for all pixels (orbit average)	20
Figure 6 : IASI NZPD determination quality flag spatial distribution (per pixel)	21
Figure 7: NZPD inter-pixel for all pixels and CCD calculated with respect to pixel 1 (orbit average)	22
Figure 8 : IASI L0 over/under-flows orbit average of all pixels	23
Figure 9 : IASI Overflows and Underflows spatial distribution (per pixel)	24
Figure 10 : Max of NZPD quality index for all pixels and CCD - BB	25
Figure 11 : Max of NZPD quality index for all pixels and CCD - CS	25
Figure 12 : Number of CSQ.	26
Figure 13 : IASI L1 data quality orbit average (% of good by PN and SB)	27
Figure 14: IASI product overall quality spatial distribution (per pixel)	28





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 6 Of: 49

Figure 15 : GQisQualIndex average (L1 data quality index for IASI sounder)	29
Figure 16 : GQisQualIndexIIS average (L1 data quality index for IASI Integrated Imager)	29
Figure 17 : GQisQualIndexSpect average (L1 data index for spectral calibration quality)	30
Figure 18 : GQisQualIndexRad average (L1 data index for radiometric calibration quality)	30
Figure 19 : GQisQualIndexLoc average (L1 data index for ground localisation quality)	31
Figure 20 : MDptPixQual average (L1 quality index for IASI integrated imager, fraction of not dead pixels)	31
Figure 21 : Instrument noise evolution between start and end of the period	32
Figure 22 : Scan mirror reflectivity evolution	33
Figure 23 : Radiometric calibration error due to scan mirror reflectivity dependency with viewing angle Maximum effect on SN1 for different scene temperature. Done with the period July 19 / May 20	34
Figure 24 : Black Body Temperature	35
Figure 25 : Focal Plane Temperature	36
Figure 26 : Monitoring of detection chain maximum delays for all bands	37
Figure 27: Ratio of calibration coefficient slopes as a function of wave number and time after the last decontamination	38
Figure 28 : Temporal evolution of calibration ratio coefficient slopes since the last decontamination. The curve was fitted with a decreasin exponential function to determine a rough date for the next decontamination (relative gain evolution of 0.8)	
Figure 29 : Contrast Monitoring	40
Figure 30 : Monitoring of interferogram baseline	41
Figure 31 : Monitoring of detection chain margins	42
Figure 32 : GFaxAxeY average (Y filtered coordinates of sounder interferometric axis)	43
Figure 33 : GFaxAxeZ average (Z filtered coordinates of sounder interferometric axis)	44
Figure 34 : Cube Corner offset variation	44
Figure 35 : Optical bench Temperature	45
Figure 36 : Column offset (black) guess vs. column offset averaged over all lines (LN) as a function of the scan position (SP=SN), and or number (along track IIS/AVHRR offset)	
Figure 37 : Line offset guess (black) vs. line offset averaged over all lines (LN) as a function of the scan position (SP=SN), and the orbit number (across track IIS/AVHRR offset)	47
Table 19: IASI TEC at CNES Toulouse	48
Table 20: EUMETCAST configuration at CNES Toulouse	48





Issue: 1.0

Date: 01/06/2020

Sheet: 7 Of: 49

LIST OF ACRONYMS

Metop-B

APO Other Parameters OPS AR Anomaly Report ASE Acquisition Start End AVHRR Advanced Very High Resolution Radiometer BB Black Body BRD BoaRD configuration CCFD Cube Corner Functional Device CCD Cube Corner Direction CCM Cube Corner Mechanism CD Cube corner Compensation Device CHART Component Health Assessment and Reporting Tool CGS Core Ground Segment at EUMETSAT CNES Centre National d'Etudes Spatiales CS Cold Space DA Applicable document DPS Data Processing Subsystem ECMWF European Centre for Medium Range Weather Forecasts EM Engineering Model EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem ISFF Instrument Point Spread Function LED Looking Filtering Device LIN Line Number		
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CD Cube corner Compensation Device CHART Component Health Assessment and Reporting Tool CGS Core Ground Segment at EUMETSAT CNES Centre National d'Etudes Spatiales CS Cold Space DA Applicable document DPS Data Processing Subsystem ECMWF European Centre for Medium Range Weather Forecasts EM Engineering Model EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	CCD	Cube Corner Direction
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CGS Contre National d'Etudes Spatiales CS Cold Space DA Applicable document DPS Data Processing Subsystem ECMWF European Centre for Medium Range Weather Forecasts EM Engineering Model EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	CD	Cube corner Compensation Device
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CS Cold Space DA Applicable document DPS Data Processing Subsystem ECMWF European Centre for Medium Range Weather Forecasts EM Engineering Model EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	CGS	Core Ground Segment at EUMETSAT
DA Applicable document DPS Data Processing Subsystem ECMWF European Centre for Medium Range Weather Forecasts EM Engineering Model EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	CNES	Centre National d'Etudes Spatiales
DPS Data Processing Subsystem ECMWF European Centre for Medium Range Weather Forecasts EM Engineering Model EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	CS	Cold Space
ECMWF European Centre for Medium Range Weather Forecasts EM Engineering Model EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	DA	Applicable document
EM Engineering Model EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	DPS	Data Processing Subsystem
EPS EUMETSAT Polar System EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	ECMWF	European Centre for Medium Range Weather Forecasts
EUMETSAT European organisation for exploitation of METeorological SATellites FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	EM	Engineering Model
FM2 / FM3 Flight Model n°2 or 3 FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	EPS	EUMETSAT Polar System
FOV Field Of View GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	EUMETSAT	European organisation for exploitation of METeorological SATellites
GRD GRounD configuration IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	FM2 / FM3	Flight Model n°2 or 3
IASI Infrared Atmospheric Sounding Interferometer IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	FOV	Field Of View
IIS Integrated Imaging Subsystem IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	GRD	GRounD configuration
IPSF Instrument Point Spread Function ISRF Instrument Spectral Response Function LFD Locking Filtering Device	IASI	Infrared Atmospheric Sounding Interferometer
ISRF Instrument Spectral Response Function LFD Locking Filtering Device	IIS	Integrated Imaging Subsystem
LFD Locking Filtering Device	IPSF	Instrument Point Spread Function
	ISRF	Instrument Spectral Response Function
IN Line Number	LFD	Locking Filtering Device
TIA THE MINING	LN	Line Number
LSB Least Significant Bit	LSB	Least Significant Bit
METOP METeorological OPerational satellite	METOP	METeorological OPerational satellite
MPF Mission Planning Facility	MPF	Mission Planning Facility
NedT Noise equivalent difference Temperature	NedT	Noise equivalent difference Temperature
NDVI Normalized Difference Vegetation Index	NDVI	Normalized Difference Vegetation Index
NZPD Number of Zero Path Difference	NZPD	Number of Zero Path Difference
ODB Operational Data Base	ODB	Operational Data Base
OPS Operational Software	OPS	Operational Software
PC Principal Component	PC	Principal Component
PDD Position Data Diagnostic	PDD	Position Data Diagnostic





Issue: 1.0

Date: 01/06/2020

Sheet: 8 Of: 49

PDU	Product Dissemination Unit
PL SOL	PayLoad Switch Off-Line
PN	Pixel Number
PTSI	Parameter Table Status Identifier
RMS	Root Mean Square
RD	Reference Document
SAA	South Atlantic Anomaly
SEU	Single Event Upset
TEC	IASI Technical Centre of Expertise (located in CNES, Toulouse)
TIGR	Thermodynamic Initial Guess Retrieval data set
VDS	Verification Data Selection
ZPD	Zero Path Difference

Metop-B





Issue: 1.0

Date: 01/06/2020

Sheet: 9 Of: 49

1 <u>INTRODUCTION</u>

The IASI TEC is based at CNES Toulouse and is responsible for the monitoring of the IASI system performances, covering both instrument and level 1 processing sub-system.

Metop-B

This document describes the activities and results obtained at the IASI TEC for instrument PFM-R on METOP-B during the following period:

Start Time: 2020/03/01
 End Time: 2020/05/31
 Orbit: 38662
 Orbit: 39968

• Duration:3 months

Note that IASI ended the Calibration / Validation (commissioning) phase on April 2013.

2 RELATED DOCUMENTS

2.1 APPLICABLE DOCUMENTS

N° Reference Titre		Titre
DA.1	IA-SP-0000-3242-CNE	Spécification de suivi de la performance en vol de IASI sur METOP-A
DA.2	IA-SP-1000-3650-CNE	Spécification de suivi par la TEC des paramètres fonctionnels de IASI en vol

2.2 REFERENCE DOCUMENTS

N°	Reference	Titre	
RD.1	EUM/OPS-EPS/REP/14/748544	Metop-B IASI annual in-flight performance report 2013	
RD.2	EUM/OPS-EPS/REP/15/796329	96329 Metop-B IASI annual in-flight performance report 2014	
RD.3	EUM/OPS-EPS/REP/16/847537	Metop-B IASI annual in-flight performance report 2015	
RD.4	EUM/OPS-EPS/REP/17/917879	Metop-B IASI annual in-flight performance report 2016	
RD.5	EUM/OPS-EPS/REP/18/993063	Metop-B IASI annual in-flight performance report 2017	





Issue: 1.0

Date: 01/06/2020

Sheet: 10 Of: 49

3 SIGNIFICANT EVENTS

The following tables present a timeline of the various requests sent by TEC and the external IASI activities.

Metop-B

Those events are typically the configuration changes, programming requests, software update, but also any external operation or activity such as mission interruption, manoeuvre, dissemination problem, ...

3.1 EXTERNAL CALIBRATION

Table 1 shows the External Calibration within the time period reported here. Note that the VDS files that come with each request are not described here.

Execution	TEC ref. ⁽¹⁾	Description	Activities	
05/03/2020 from 5h15 to 9h11 orb. 38721 to 38723	RM-87			
03/04/2020 from 5h15 to 9h11 orb. 39133 to 39135	RM-88		For routine monitoring (IIS and IASI NeDT,	
02/05/2020 from 5h15 to 9h11 orb. 39545 to 39547	RM-89		scan mirror reflectivity, ghost,)	
31/05/2020 from 5h15 to 9h11 orb. 39957 to 39959	om 5h15 to 9h11 RM-90			

Table 1: External Calibration TEC Requests

- · a TEC request or
- a "MPF" uploaded directly by EUMETSAT in full accordance with TEC. The reference "Monthly_MPF" is based on the March 2008 TEC External Calibration request. The MPF for moon avoidance is based on the December 2008 TEC External Calibration request: "ICAL_OCF_xx_M02_20081216060000Z_20090616060000Z_20081209100934Z_IAST_EXTCALIBRA.dts"

3.2 ON BOARD CONFIGURATION

On-board processing configuration updates that had been made within the time period reported here:

PTSI	IASI on board parameter files	Delivery by TEC	activated on	TEC ref.	affected parameters of a DPS TOP configuration update

Table 2: DPS and MAS configuration TEC Requests

Delivery applicable at the beginning of the period:

PTSI	IASI on board parameter files	Delivery by TEC	activated on	TEC ref.	affected parameters of a DPS TOP configuration update
17 1.0	IDPS_OBP_xx_M01_20191106000000 Z_20200506000000Z_20191105145844 Z_IAST_DPSPARAMOD.tar	05/11/19	10/12/2019, orbit 37504	R_50	Update of Reduced Spectra

Table 3: DPS and MAS previous configuration

The associated ground configuration table (BRD file), necessary to handle coherent configuration at system level, is presented in the next section. These associated configuration table are necessary for L1 processing.

⁽¹⁾ TEC convention: R for Routine, M for Monthly and L for moon avoidance, followed by a chronological number

⁽²⁾ An external calibration could be the result of:





Issue: 1.0

Date: 01/06/2020

Sheet: 11 Of: 49

3.3 GROUND CONFIGURATIONS UPDATES FOR LEVEL 1 PROCESSING

Metop-B

On-ground processing configuration updates that had been made within the time period reported here:

IDef	IASI L1 auxiliary files	Delivery by TEC	Upload on GS1	Content

Table 4: IASI L1 Auxiliary File Configuration on the Operational EPS Ground Segment

Delivery applicable at the beginning of the period:

IDef	IASI L1 auxiliary files Delivery by TEC		Upload on GS1	Content
50	50 IASI_BRD_xx_M01_20191106000000Z_xxx		BRD activated on 10/12/2019 13:06, orbit 37504	Update of Reduced Spectra
23	IASI_GRD_xx_M01_20190628000000Z_xxx xxxxxxxxxxZ_20190627074600Z_IAST_00 00000023	27/06/19	GRD activated on 30/07/2019 08:17, orbit 35612	Update of scan mirror reflectivity
12	12 IASI_ODB_xx_M01_20130417100000Z_xxx xxxxxxxxxxZ_20130417082506Z_IAST_00 00000012		ODB activated on 14/05/13 12:05 orbit 3393	

Table 5: IASI L1 auxiliary file previous configuration

3.4 DATA BASES UPDATE FOR THE USERS

The Noise Covariance Matrix (NCM) and Spectral data base (SDB) are specific data bases for the users. They are updated according to the main ground level 1 evolutions.

Updates of the NCM and SDB that had been made within the time period reported here:

IDef	Users Data-Base	Delivery by TEC	TEC ref.	Comments

Table 6: IASI Data Bases for the users

Delivery applicable at the beginning of the period:

IDef	Users Data-Base	Delivery by TEC	TEC ref.	Comments
2	IASI_NCM_xx_M01_20130522081244Z_ 20130522081244Z_20130522081245Z_IA ST_SPECTRESPO	22/05/13	CVA_COV_2	Update of NCM
12	SI_SDB_xx_M01_20130923150000Z_ 130923150000Z_20130923132439Z_IA		User database associated to ODB IDefSDB 12	

Table 7: previous IASI Data Bases





Issue: 1.0

Date: 01/06/2020

Sheet: 12 Of: 49

3.5 ON GROUND HW/SW EVOLUTION

Updates of PPF L1 software within the time period reported here:

IASI L1 PPF software version	Delivery by TEC	Date introduced on GS1	Comments

Metop-B

Table 8: IASI L1 PPF Configuration on the Operational EPS Ground Segment

Software version applicable at the beginning of the period:

IASI L1 PPF software version	Delivery by TEC	Date introduced on GS1	Comments
8.2	19/09/2019	24/09/2019 for sensing time 08:59 Orbit 36408	Flight dynamics anomaly with MetopLib

Table 9: Previous IASI L1 PPF

3.6 DECONTAMINATION

Decontaminations that have been made or requested within the time period reported here:

Last due date	Date of decontamination	Description			

Table 10: Decontamination TEC Requests

Previous decontamination:

Last due date	Date of decontamination	Description
End June 2018	27 June to 2 July 2018	Decontamination 200K

Table 11: Previous decontamination

3.7 INSTRUMENT

3.7.1 External events

This category is for those activities/events that are external to IASI but still have an impact. It is broken down into classes of *PL-SOL* and *OOP* manoeuvre.

3.7.1.1 Manoeuvres

Date	Type(*)	Description	IP flag	OoP mission Outage
13/05/2020	IP	IP manoeuvre #37 (orbit 39705)	Yes	

Table 12: Overview of METOP manoeuvres in the reporting period

^{(*):} IP for In-Plane manoeuvres (IASI stays in NOp) and OoP for Out of plane manoeuvres (IASI is put in Heater 2)





Issue: 1.0

Date: 01/06/2020

Sheet: 13 Of: 49

3.7.1.2 PL-SOL

PL-SOL events that have occurred within the time period reported here:

Dates	Orbits	Description

Metop-B

Table 13: PL-SOL

3.7.2 Operation leading to mission outage

This chapter presents the intervention on IASI needing routine interruption that have occurred within the time period reported here.

Dates	Orbits	type	IASI mode	Description

Table 14: Scheduled interruptions

3.7.3 Anomaly leading to mission outage

Tables 15 and 16 present the major and minor anomalies internal to IASI that have occurred within the time period reported here.

Note that in this section minor anomalies are all identified and without any impact on the mission, and major anomalies only affect IASI instrument, and no other sub-systems of the spacecraft.

Dates	Orbits	Anomaly type (*)	IASI mode	Description
27/03 23:39 to 28/03 03:50	39044 to 39047	SSR		Loss of X-band data due to SSR anomaly
27/04 14:56	39480 DPS			DPS Auto-recovery function triggering SEU in DPS subsystem (DPS Pixel mode Acquisition)

Table 15: Major anomalies

(*): SEU (LAS, CCM or DPS) anomalies or SET anomalies

Day	Orbits	error n°	Severity	Anomaly type	LN	SN	Description
28/04/2020 05:26	39488	8	1	SQ2	5225	30	CCD=1 (forward)

Table 16: Minor anomalies





Issue: 1.0

Date: 01/06/2020

Sheet: 14 Of: 49

4 PERFORMANCE MONITORING

4.1 PERFORMANCE MONITORING

In order to ensure that the IASI system is permanently running in good conditions, the CNES (IASI TEC) and EUMETSAT (CGS) are monitoring products at various temporal levels: at line, PDUs and DUMP (full orbit).

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The on-board and ground processing performance algorithms issue more than one hundred quality indicators, called flags and simple parameters. Those are alarms for any bad functioning or local performance degradation.

According to the results, the TEC is also in charge of delivering new on-board or ground parameters to EUMETSAT when it is necessary. EUMETSAT is then in charge of uploading them on-board or as an input of the level 1 processing chain. During the whole instrument life, these parameter adjustments are necessary in order to take into account instrument evolution in the processing and finally to maintain a good data quality.

Colour code used for the status report.

Status Colour	Meaning		
GREEN	≥ 95% of good spectra		
YELLOW	< 95% of good spectra		
RED	Production interrupted		
BLANK	No Status Reported		

Table 17: Functional status legend





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 15 Of: 49

4.2 PERFORMANCE SYNOPSIS

Synthetic view of all the indicators evaluated for L0/L1 data and their current status.

Section	Component	Description	Status	Comments
4.3	LO	Level-0 Data Quality Overall quality Main flag and quality indicator parameters Spikes monitoring ZPD monitoring Overflows/Underflows monitoring Reduced Spectra monitoring Second level flag and quality indicators	GREEN	On-board processing
4.4	L1	 Level-1 Data Quality Overall quality Main flag and quality indicator parameters 	GREEN	On ground processing
<u>4.5</u>	L1	 Sounder radiometric performances Radiometric noise Radiometric calibration Acquisition chain delay Optical transmission Ice Prediction of decontamination date Interferometric contrast Interferogram Baseline Detection chain 	GREEN	
4.6	L1	Sounder spectral performances Dimensional stability Position of axis Cube Corner constant offset Cube Corner velocity Optical bench temperature Spectral calibration	GREEN	
4.7	L1	Geometric performances Sounder/IIS co-registration IIS/AVHRR co registration	GREEN	
4.8	L1	IIS radiometric performances	GREEN	

Table 18: IASI product components functional status





Issue: 1.0

Doc n°:

Date: 01/06/2020

Sheet: 16 Of: 49

IA-RP-2000-4510-CNE

4.3 LEVEL 0 DATA QUALITY (L0)

4.3.1 Overall quality

The IASI L0 data quality (orbit average) through IASI engineering products is shown in Figure 1.

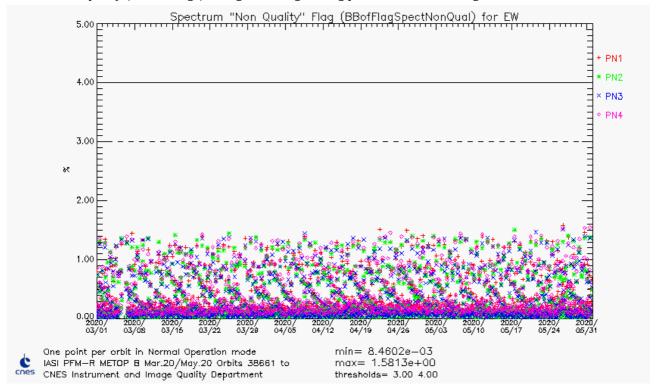


Figure 1: IASI L0 data quality orbit average (per pixel and CCD)

The geographical distribution of the overall L0 (board) quality flag for the 4 pixels is shown in Figure 2.





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 17 Of: 49

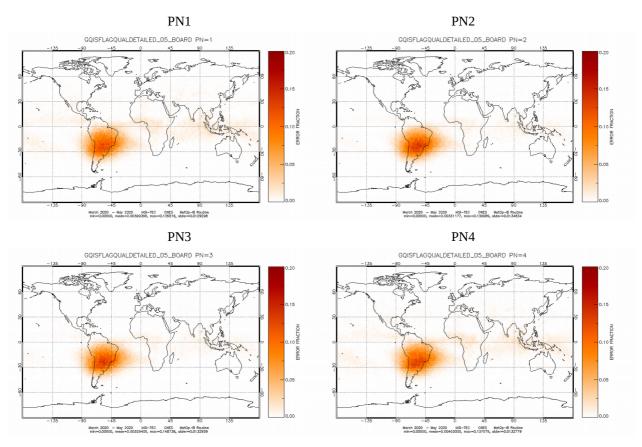


Figure 2 : IASI L0 data quality spatial distribution (per pixel)

The IASI L0 quality and on-board processing are nominal.





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 18 Of: 49

4.3.2 Main flag and quality indicator parameters

The main contributors to the rejected spectra by on-board processing are: spikes (proton interaction on detectors), failure of NZPD algorithm determination and over/underflows (measured data exceeding on-board coding tables capacity). They are analysed in details hereafter.

4.3.2.1 Spikes monitoring

Spikes occur when a proton hits a detector. This very high energetic particle disrupts the measure of the interferogram and then corrupts the spectrum.

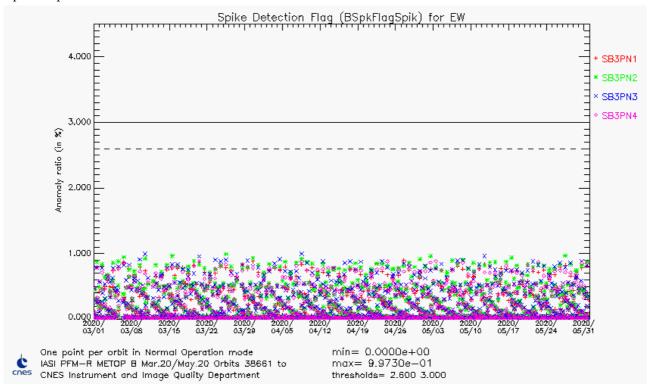


Figure 3: Temporal evolution of spikes anomaly ratio in % for all pixels (orbit average)

An example of the geographical distribution of spikes occurrences on band 3 for the 4 pixels is shown in Figure 4.





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 19 Of: 49

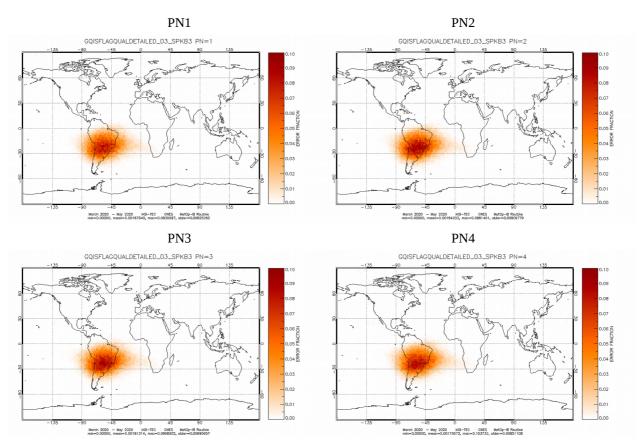


Figure 4: Geographical distribution of spikes occurrences in % for band 3 and all pixels

Spikes are mainly located in the regions of Earth where the magnetic field doesn't protect the satellite from the energetic particles: the poles and the SAA (South Atlantic anomaly).

Spike anomaly ratio is nominal for the reported period.





Issue: 1.0

Date: 01/06/2020

Sheet: 20 Of: 49

4.3.2.2 ZPD monitoring

The ZPD ("Zero Path Difference") is the position of the central fringe of the interferogram. The NZPD is the number of the sample detected as the ZPD. On IASI, it is determined by a software. This is a special feature of IASI in comparison to other instruments for which NZPD determination is done by hardware.

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NZPD variations are governed by two phenomenons:

- 1. ASE fluctuations which have the same effect on each pixel and can produce NZPD variation of 30-40 samples over month. This is the first order phenomena.
- 2. Mechanical deformation of the interferometer or evolution of detection chain delays. These phenomenons affect the 4 pixels in different way. However this phenomenon has a second order effect in comparison to the first one.

We monitor both NZPD determination quality flag and interpixel homogeneity. We expect stability.

BZPDFlagNZPDNonQualEW: Temporal evolution of NZPD determination quality flag for earth view

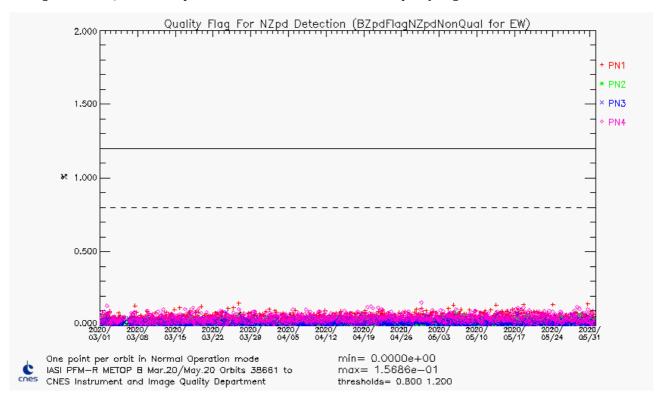


Figure 5: Temporal evolution of NZPD determination anomaly ratio in % for all pixels (orbit average)

NZPD determination anomaly ratio is nominal for the reported period.

The geographical distribution of the NZPD determination quality flag for the 4 pixels is shown in Figure 6.





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 21 Of: 49

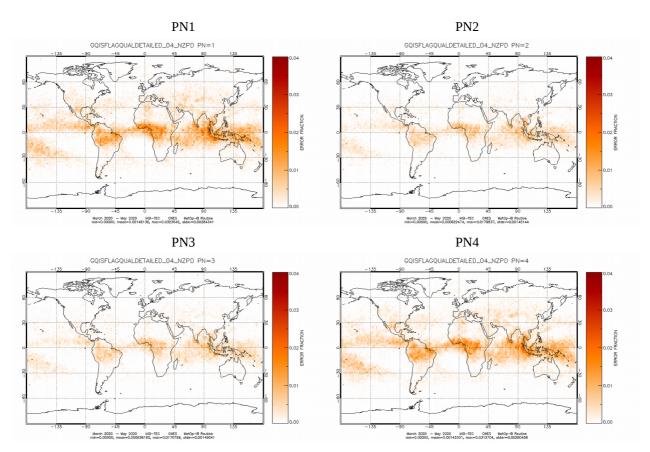


Figure 6: IASI NZPD determination quality flag spatial distribution (per pixel)

The NZPD determination fails over some clouds that have a temperature that induces no energy in the central fringe of the interferogram.





Issue: 1.0

Date: 01/06/2020

Sheet: 22 Of: 49

NZPD inter-pixel homogeneity monitoring

This monitoring is necessary in order to follow potential deformation of the interferometer or evolution of detection chain delay.

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The NZPD inter-pixel homogeneity is nominal over the reported period. Consequently, these parameters are perfectively stable and in-line with the specification.

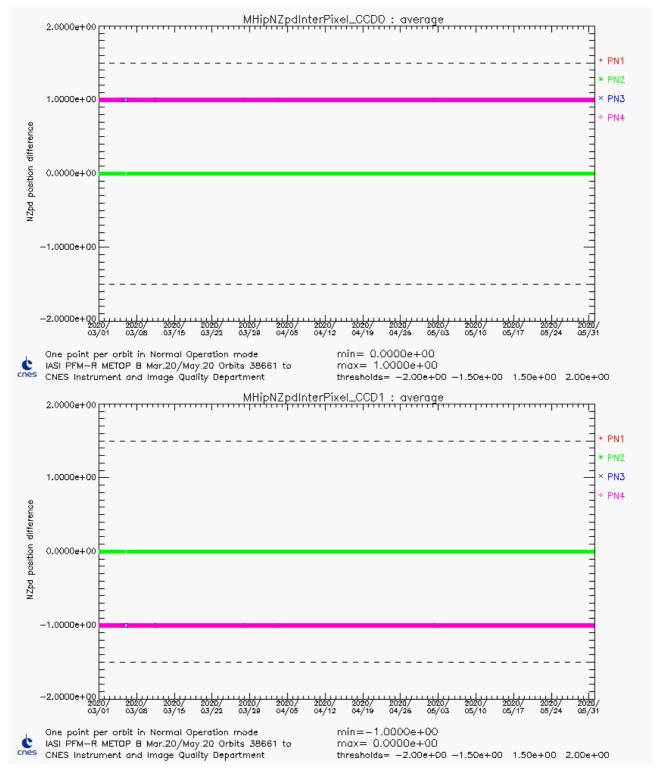


Figure 7: NZPD inter-pixel for all pixels and CCD calculated with respect to pixel 1 (orbit average)





Issue: 1.0

Date: 01/06/2020

Sheet: 23 Of: 49

4.3.2.3 Overflows / Underflows monitoring

The total number of bits available for a spectrum to be transmitted to the ground is limited. For that reason, we have defined coding tables to encode each measured spectrum. These tables have been designed by using "extreme spectrum" corresponding to known drastic atmospheric conditions. The coding step is also set to not introduce additional noise into the spectrum. However for very extreme atmospheric conditions (sunglint in B3, very high stratospheric temperature...) a measurement can exceed on-board coding tables' capacity and causes an over/underflow.

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Over/underflows occurrences are monitored and stability is expected. As long as they remain to low levels, the coding table is not changed. Note that changing the coding tables requires compromises. Indeed, increasing the encoding capacity can be achieved by two different ways. A first solution consists in an increase of the coding step without changing the number of bits. However, that leads to an increase of the digitalization noise. Then, a second solution consists in keeping the coding step constant while increasing the number of bits available for a particular band. But, the total amount of bits available for the entire spectrum is limited and constant. So, that requires to decrease the encoding capacity in another spectral band.

Time series of Overflows and Underflows (orbit average) are shown in following figure for all pixels.

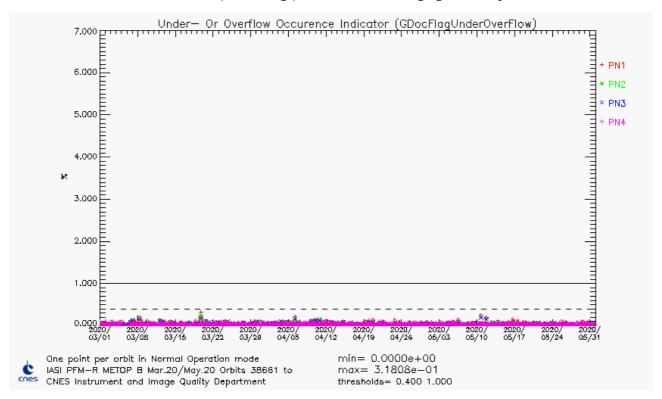


Figure 8: IASI L0 over/under-flows orbit average of all pixels

Over/underflows ratio is nominal for the reported period.

The geographical distribution of the Overflows and Underflows for the 4 pixels is shown in Figure 9.





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 24 Of: 49

PN1

COISFLAGQUALDETAILED_OB_DVINDER PN=1

OOSFLAGQUALDETAILED_OB_DVINDER PN=2

OOSFLAGQUALDETAILED_OB_DVINDER PN=3

OOSFLAGQUALDETAILED_OB_DVINDER PN=3

OOSFLAGQUALDETAILED_OB_DVINDER PN=4

OOSFL

Figure 9: IASI Overflows and Underflows spatial distribution (per pixel)

4.3.2.4 Reduced Spectra monitoring

On-board Reduced Spectra is one of the most important parameter to monitor. It ensures that on-board spectra still have a good radiometric calibration when on-board configuration reduced spectra are reloaded. This is the case, for instance, after an instrument mode change.

Reduced spectra are slightly evolving with respect to potential deformation of the interferometer (optical bench).

In order to prevent from a large difference between current and on-board configuration reduced spectra, we apply the DPS processing on the verification interferograms using the reduced spectra from the on-board configuration (TOP) instead of the filtered reduced spectra computed on-board with the current calibration views. These reduced spectra from the on-board configuration are used as initialisation each time there is mode change. If they are too far from the reality, no spectra can be computed on-board after a mode change. We monitor the evolution of ZPD determination quality index for calibration views (BZpdNzpdQualIndexBB and CS) obtained by this DPS processing at TEC, results of this monitoring are given hereafter.





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 25 Of: 49

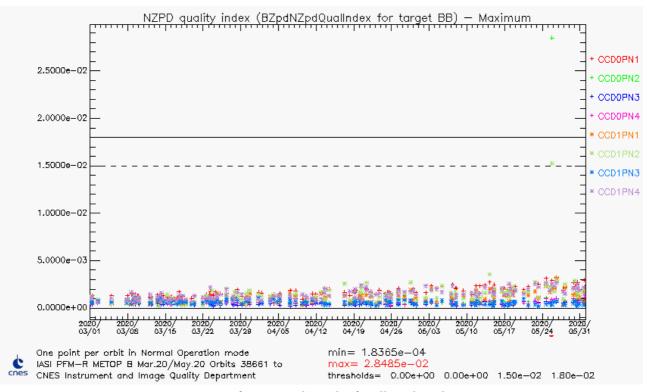


Figure 10: Max of NZPD quality index for all pixels and CCD - BB

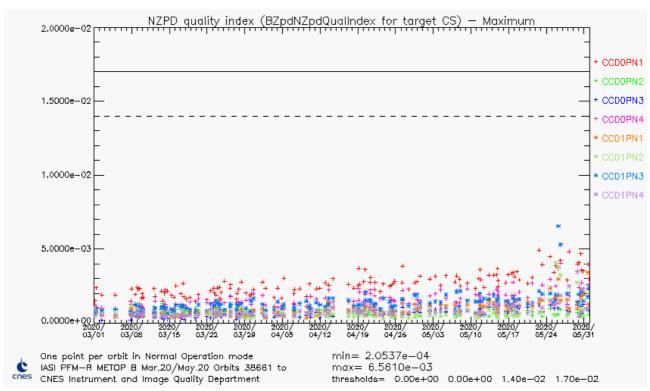


Figure 11: Max of NZPD quality index for all pixels and CCD - CS

The high value for target BB on May 25th is probably due to micro-vibrations (suspected SEU on MHS instrument).

As soon as BZPDNZPDQualIndexBB and CS remain below 0.02 on-board reduced spectra are robust to an instrument mode change. The reduced spectra quality is well within specification since the last update of the on-board reduced spectra performed in December 2019.





Issue: 1.0

Date:

Metop-B

01/06/2020

Sheet:

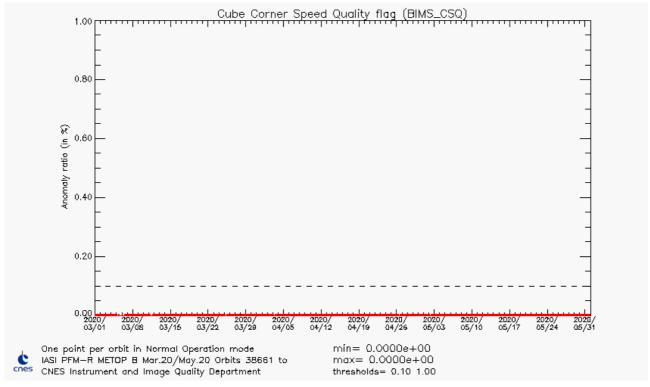
26

49

Of:

4.3.2.5 Cube corner Speed Quality (CSQ) monitoring

From verification products (BB & CS):



From engineering products (EW):

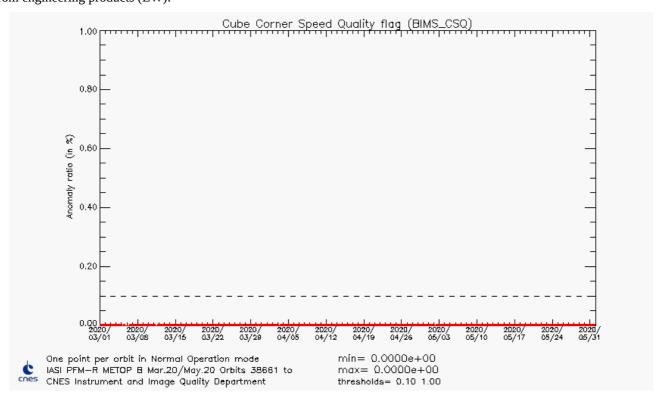


Figure 12: Number of CSQ





Issue: 1.0

Date: 01/06/2020

Sheet: 27

Of: 49

4.3.3 Second level flags and quality indicators

All second level flags and indicators are stable and nominal.

4.3.4 Conclusion

L0 Flag and quality indicators are stable.

4.4 LEVEL 1 DATA QUALITY (L1)

4.4.1 Overall quality

The IASI overall quality is shown as the orbit averages of the quality indicator for the individual pixels in the next figure.

Metop-B

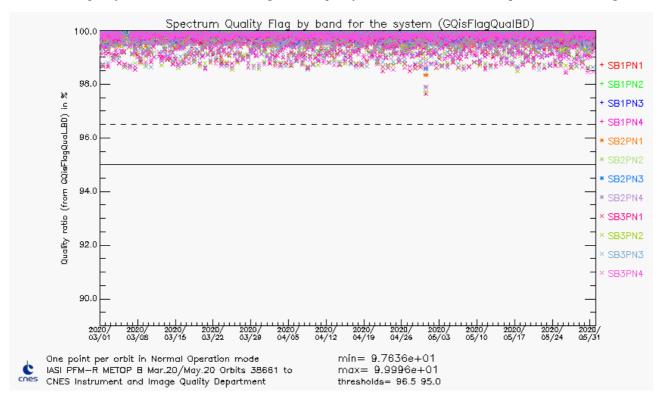


Figure 13: IASI L1 data quality orbit average (% of good by PN and SB)

One should note that, over the period covered by the present document, the averaged data rejection ratio is less than 1%. We clearly see that data quality is better on the bands B1 and B2 in comparison to band B3 (which is the most affected by spikes).

The geographical distribution of the IASI product overall quality for the 4 pixels is shown in Figure 14.





Doc n°:

IA-RP-2000-4510-CNE

1.0 Issue:

01/06/2020 Date:

Sheet: 28 Of: 49

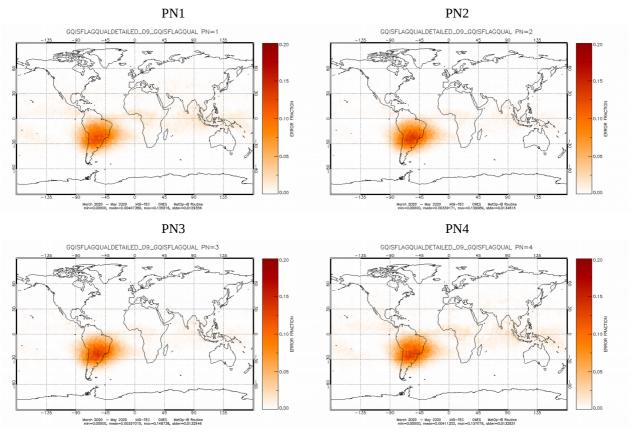


Figure 14: IASI product overall quality spatial distribution (per pixel)

The main contributors are the spikes (mainly in band 3, which is the band the most sensitive to the spikes).

4.4.2 **Main flag and quality indicator parameters**

All the quality indexes that follow are general L1 quality indexes of sounder products.

GQisQualIndex – average – is the average general quality index of the sounder products.

GQisQualIndexIIS is the IASI integrated imager (IIS) images quality index.

GQisQualIndexSpect is the spectral quality index of the sounder products.

GQisQualIndexRad is the radiometric quality index of the sounder products.

GqisQualIndexLoc is the ground localisation quality index of the sounder products.

MDptPixQual is a quality index for IASI integrated imager (IIS) that represents a fraction of not dead pixels.





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 29 Of: 49

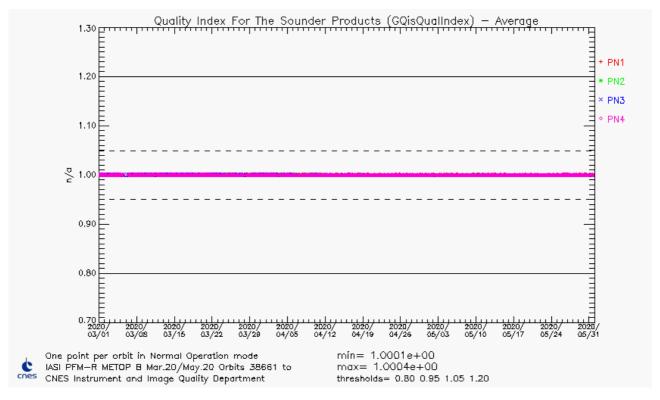


Figure 15 : GQisQualIndex average (L1 data quality index for IASI sounder)

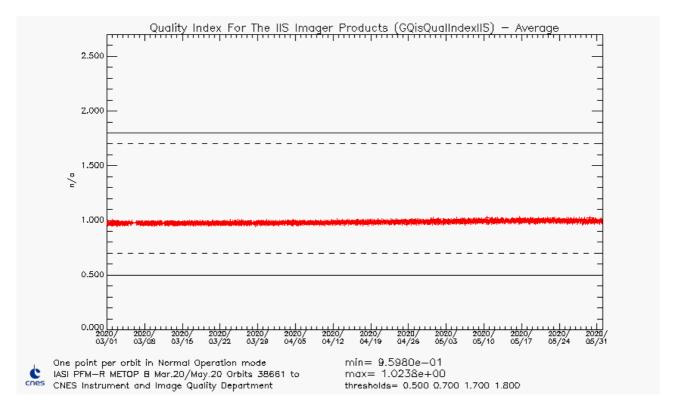


Figure 16: GQisQualIndexIIS average (L1 data quality index for IASI Integrated Imager)





Doc n°: IA-RP-2000-4510-CNE Issue: 1.0

Date: 01/06/2020

Sheet: 30 Of: 49

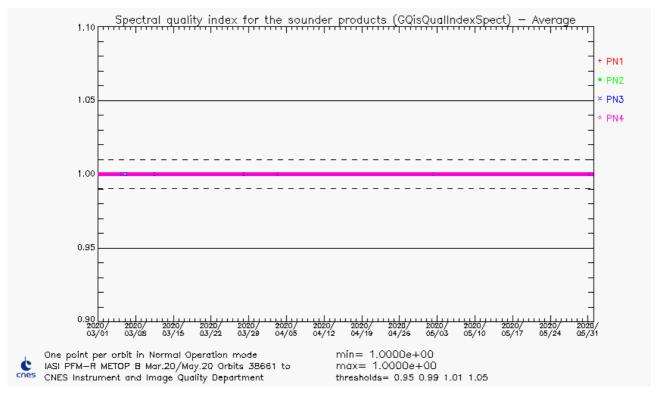


Figure 17: GQisQualIndexSpect average (L1 data index for spectral calibration quality)

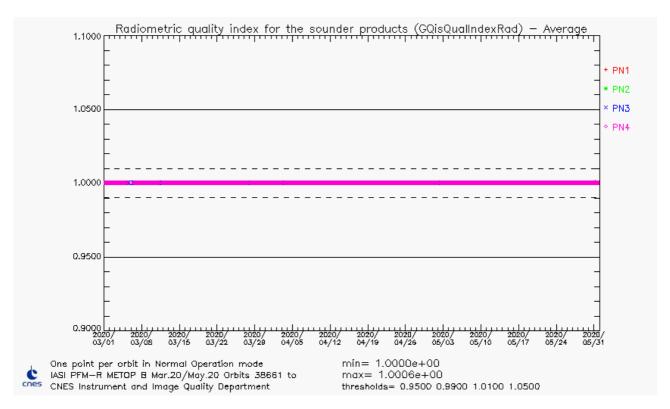


Figure 18: GQisQualIndexRad average (L1 data index for radiometric calibration quality)





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 31

Of: 49

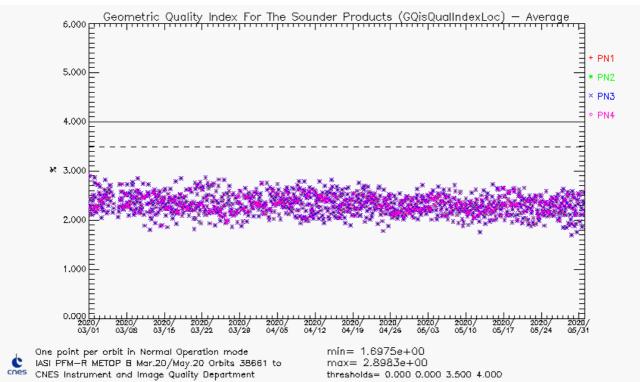


Figure 19: GQisQualIndexLoc average (L1 data index for ground localisation quality)

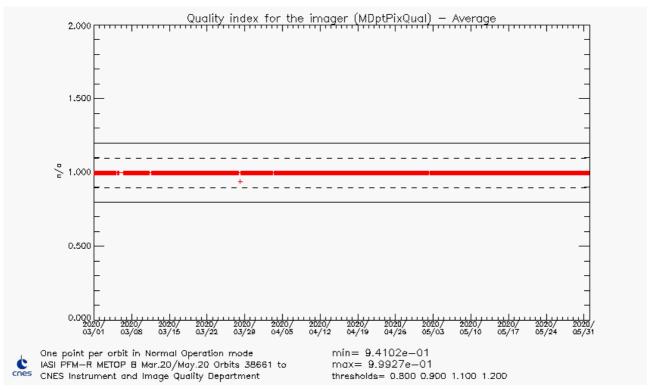


Figure 20: MDptPixQual average (L1 quality index for IASI integrated imager, fraction of not dead pixels)

4.4.3 Conclusion

L1 Flag and quality indicators are stable and meet the specifications.





Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 32 Of: 49

4.5 SOUNDER RADIOMETRIC PERFORMANCES

4.5.1 Radiometric Noise

Monitoring the radiometric noise allows to monitor the long term degradation of the instrument as well as to look for punctual anomaly of IASI or other component of METOP.

Monthly L0 noise estimation (CE)

This monthly estimation is performed during routine External Calibration on BB views.

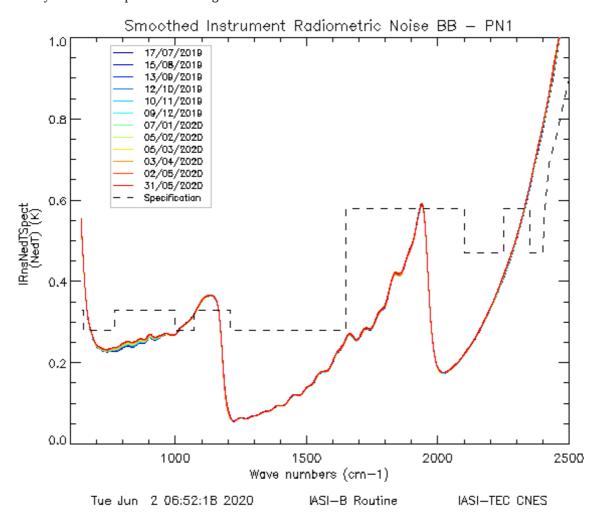


Figure 21: Instrument noise evolution between start and end of the period

The instrument noise is very stable apart from ice effect, especially between 700 and 1000 cm⁻¹. This point will be developed in section 4.5.4.1.





Issue: 1.0

Date: 01/06/2020

Sheet: 33 Of: 49

4.5.2 Radiometric Calibration

The radiometric calibration allows to convert an instrumental measurement into a physical value. The radiometric calibration is used to convert an interferogram into an absolute energy flux by taking into account instrument discrepancies. Even if the calibration has been studied on ground, it has to be continuously monitored in-flight in order to follow any potential degradation of the instrument (optics, detectors ...).

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<u>Approach</u>: Radiometric fine characterization has been done during on-ground testing and Cal/Val. All parameters likely to cause a failure in radiometric calibration process have been identified and are continuously monitored. As long as they remain stable, there is no problem with radiometric calibration.

Evolution of scanning mirror reflectivity

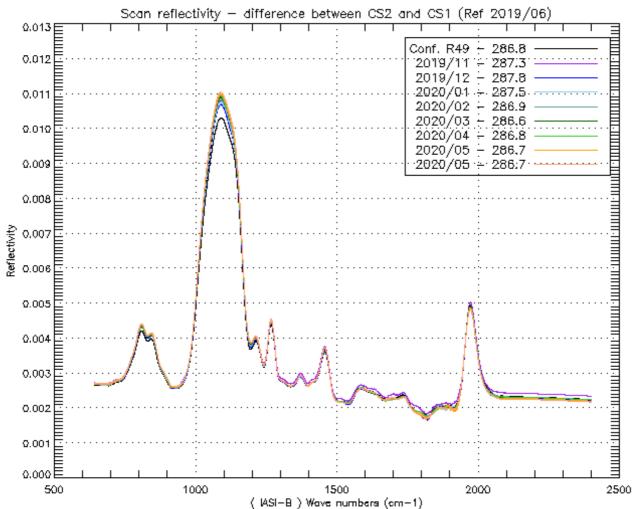


Figure 22: Scan mirror reflectivity evolution

The reference reflexivity (in black) is the one computed on external calibration data from June 18th 2019. We see a slight evolution within [1000-1100 cm⁻¹] band. Values for wavenumbers greater than 2400 cm⁻¹ are not significant because of instrument noise.





Issue: 1.0

Date: 01/06/2020

Sheet: 34 Of: 49

The next figure shows the translation of scan mirror reflectivity in terms of maximum radiometric calibration error for different scene temperatures.

Metop-B

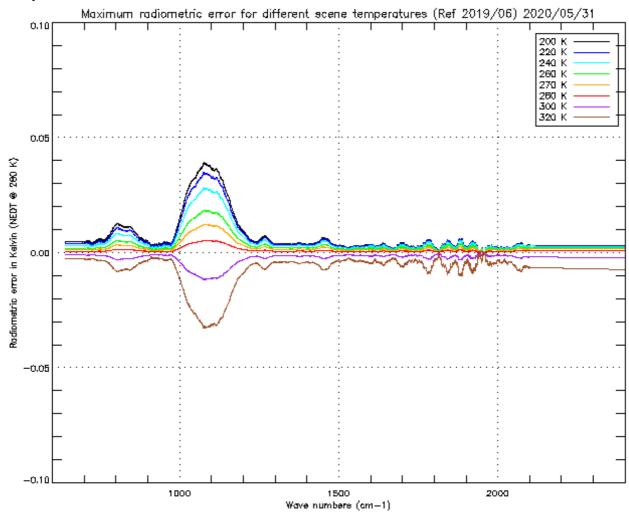


Figure 23: Radiometric calibration error due to scan mirror reflectivity dependency with viewing angle Maximum effect on SN1 for different scene temperature.

Done with the period July 19 / May 20

In any cases radiometric calibration maximum error is lower than the specification (0.1K). The scan mirror reflectivity law (on ground configuration), prepared with June 18^{th} routine External Calibration data, has been updated in the operational ground segment on July 30^{th} 2019.





Doc n°: I

IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 35

Of: 49

Internal black body

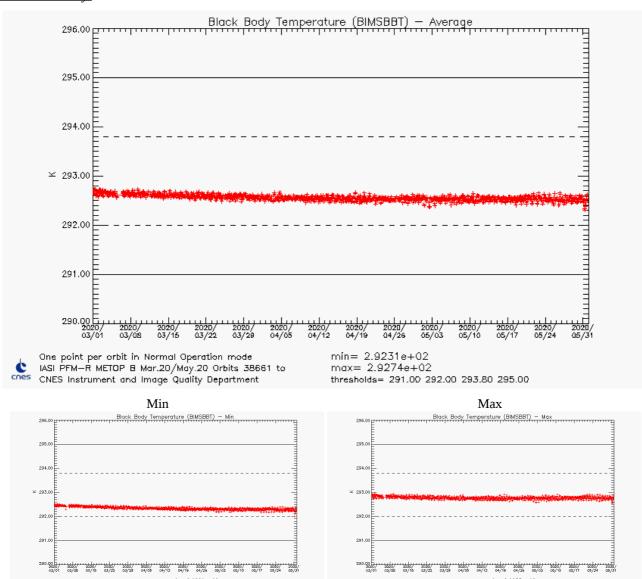


Figure 24: Black Body Temperature

The black body temperature is stable.





Issue: 1.0

Date: 01/06/2020

Sheet: 36 Of: 49

Non linearity of the detection chains

Non-linearity tables of the detection chains are still nominal as long as sounder focal plane temperature variation amplitude is lower than 1K.

Metop-B

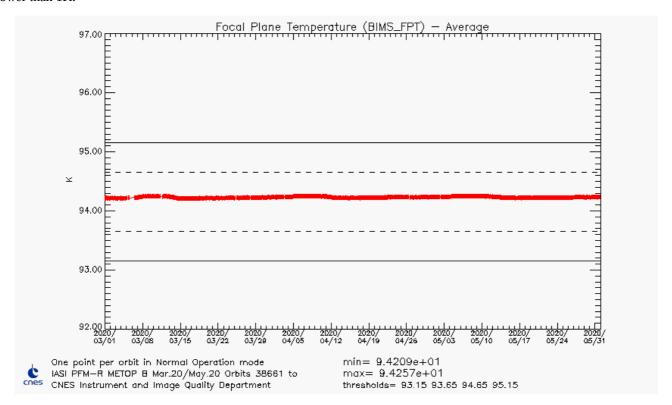
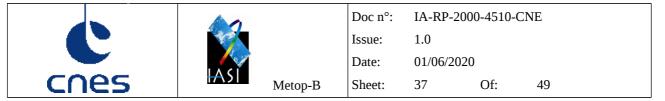


Figure 25 : Focal Plane Temperature



4.5.3 **Delay of Detection Chains**

Long term stability and values lower than 400 ns are required in order to properly take into account cube corner velocity fluctuations.

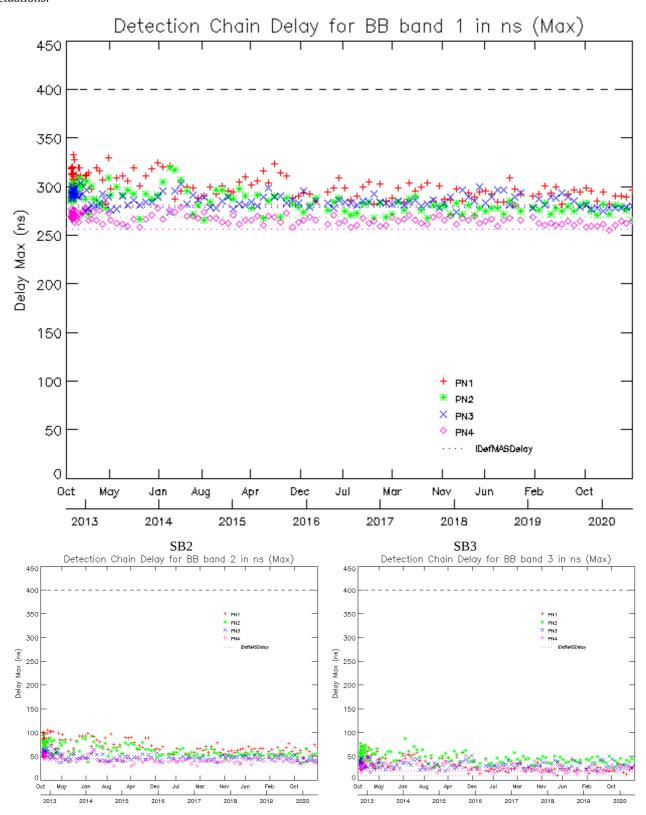


Figure 26: Monitoring of detection chain maximum delays for all bands





Issue: 1.0

Date: 01/06/2020

Sheet: 38 Of: 49

4.5.4 Optical Transmission

4.5.4.1 Ice

The IASI interferometer and optical bench are regulated at 20°C temperature, while the cold box containing cold optics and detection subsystem is at about -180°C. Water desorption from the instrument causes ice formation on the field lens at the entrance of IASI cold box. This desorption phenomenon is particularly important at the beginning of the instrument in-orbit life. That's why one of the very first activities of IASI in-orbit commissioning was an outgasing phase consisting in heating the cold box up to 300 K during 20 days (from 22th September 2012 until 16th October 2012). This operation allows removing most of the initial contaminants coming from IASI and other MetOp instruments. A routine outgasing is then needed from time to time to remove ice contamination, but less and less frequently as the desorption process becomes slower.

The first routine outgassing procedure (shorter duration and at 200K) was done from 10th to 14th March 2014.

Metop-B

The second one was done from 25th to 30th November 2015.

The third one was done from 27th June to 2nd July 2018.

The maximum acceptable degradation of transmission is about 20% loss at 850 cm $^{-1}$ (which corresponds to an ice deposit thickness of about 0.5 μ m).

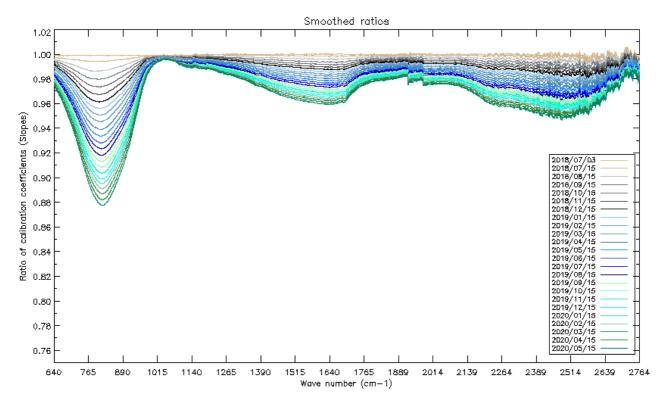


Figure 27 : Ratio of calibration coefficient slopes as a function of wave number and time after the last decontamination





Issu

IA-RP-2000-4510-CNE

Of:

Issue: 1.0

Doc n°:

Date: 01/06/2020

Sheet: 39

49

4.5.4.2 Prediction of decontamination date

The transmission degradation rate is regularly monitored by CNES TEC through gain measurements given by calibration coefficients ratios.

Metop-B

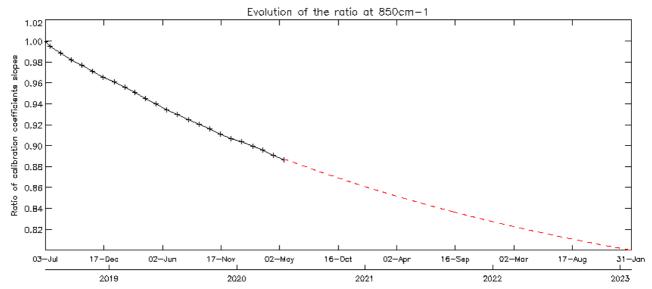


Figure 28: Temporal evolution of calibration ratio coefficient slopes since the last decontamination. The curve was fitted with a decreasing exponential function to determine a rough date for the next decontamination (relative gain evolution of 0.8)

The loss of instrument gain due to ice contamination is, as expected, decreasing over time. The next decontamination is not expected before beginning of 2023.





Metop-B

Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 40 Of: 49

4.5.5 Interferometric Contrast

The interferometric contrast is defined as the interferogram fringe discrimination power. Figure 29 shows temporal evolution of instrument contrast since the beginning of IASI life in orbit for all pixels and all CCD.

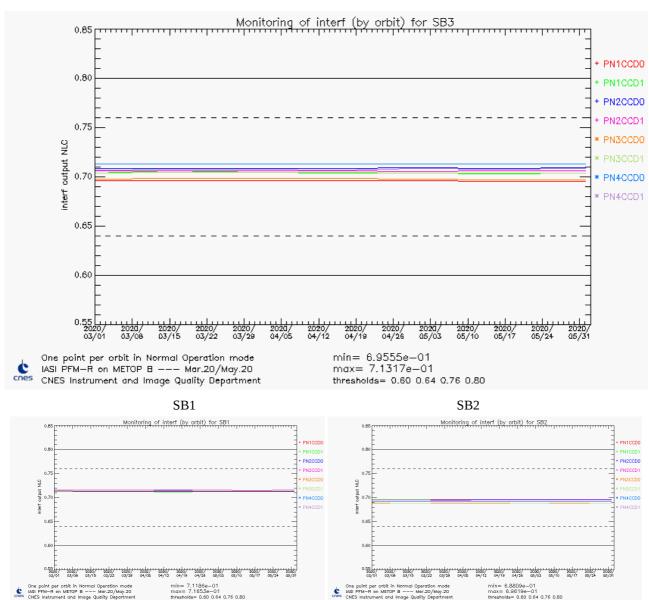


Figure 29: Contrast Monitoring





Issue: 1.0

Date: 01/06/2020

Sheet: 41 Of: 49

4.5.6 <u>Interferogram Baseline</u>

The interferogram baseline is the mean value of the interferogram. Figure 30 shows temporal evolution of the baseline of the raw interferograms on calibration targets (BB and CS). The values are raw values, they are not physical, but the evolution is interesting: as the values are proportional to the energy received from a target and calibration targets are stable, the evolution can show the decrease of instrument transmission or events due to energetic particles.

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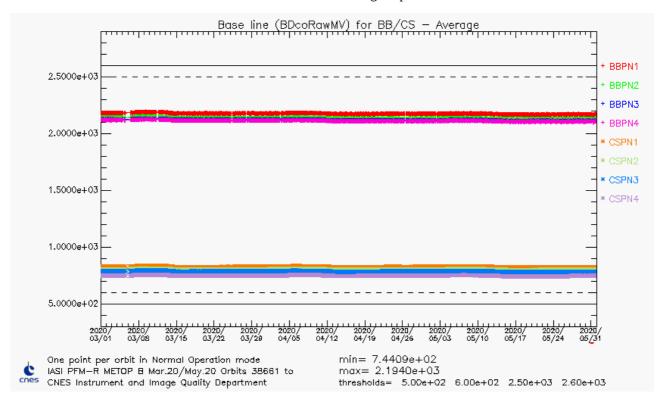


Figure 30: Monitoring of interferogram baseline





Issue: 1.0

Date: 01/06/2020

Sheet: 42 Of: 49

4.5.7 **Detection Chain**

Detection chains are tuned in gain and offset via telecommand. The goal is to avoid saturation while conserving the maximum dynamic to limit digitalization noise.

Metop-B

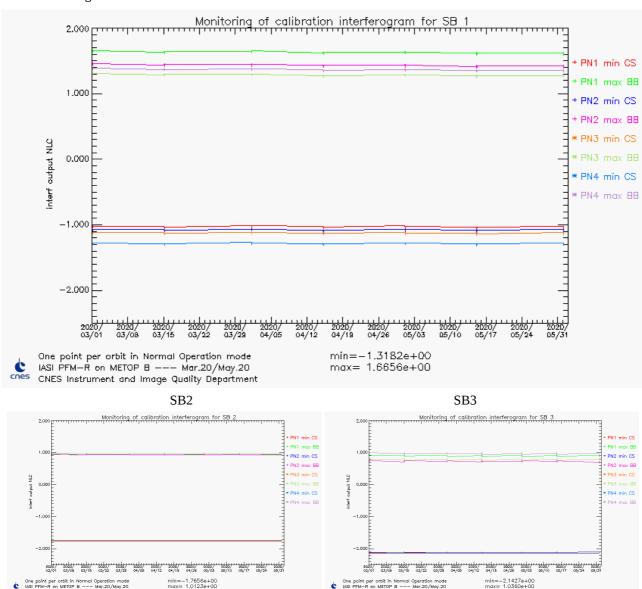


Figure 31: Monitoring of detection chain margins

Margins are sufficient for the moment.

4.5.8 Conclusion

The radiometric performances of IASI are nominal and stable.

An extrapolation of the current calibration ratio leads to a rough date for the next decontamination at the beginning of 2023. Scan mirror reflectivity was updated end July 2019. The next update is not foreseen before end 2020.





Issue: 1.0

Date: 01/06/2020

Sheet: 43 Of: 49

4.6 SOUNDER SPECTRAL PERFORMANCES

The goal of the spectral calibration is to provide the best estimates of spectral position of the 8461 IASI channels.

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The large sensitivity of infrared spectrum to spectral calibration errors has led to stringent specifications:

- A prior knowledge of spectral position better than of 2.10⁻⁴ (design)
- A posterior maximum spectral calibration relative error of 2.10⁻⁶ (after calibration by OPS)

In order to reach the specification of 2.10⁻⁶, we need an accurate Instrument Spectral Response Function (ISRF) model. This model have been done and validated in the early time of IASI development.

For sake of operational time constrain, complete ISRF calculation is not done in real-time by OPS software but pre-calculated and stored in a database called "spectral database". OPS processing determine on-line the most relevant instrument function to be used by OPS with respect to current values of a set of parameters (interferometric axis, cube corner offset…).

The approach to monitor IASI spectral performances is very similar to the one used for radiometric calibration. Spectral calibration fine characterization has been done during ground testing and Cal/Val. All parameters likely to cause a failure in spectral calibration process have been identified and are continuously monitored. As long as they remain stable, there is no problem with IASI spectral calibration.

In addition, a spectral calibration assessment is done over homogeneous scenes when IASI is in external calibration, nadir view

Since the permanent compensation device stop on 7th October 2015, the monitoring of the ghost was cancelled as the data are no more affected by the ghost.

4.6.1 Monitoring of the ISRF inputs

4.6.1.1 Position of the interferometric axis

The interferometric axis is the cube corner displacement direction. Its value has changed several times during CalVal due to the various configurations used. Since the end of CalVal, its value is now stable around ($Y = 1060\mu rad$; $Z = 1210\mu rad$). The central position used in the "spectral database" generation, are $1000\mu rad$ and $1200\mu rad$, respectively for Y and Z axis.

Since the drift of the interferometer axis is lower than 300 urad, there is no need to update the "spectral database".

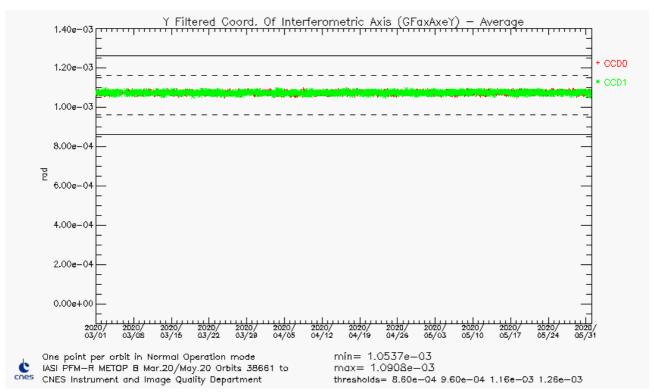


Figure 32: GFaxAxeY average (Y filtered coordinates of sounder interferometric axis)





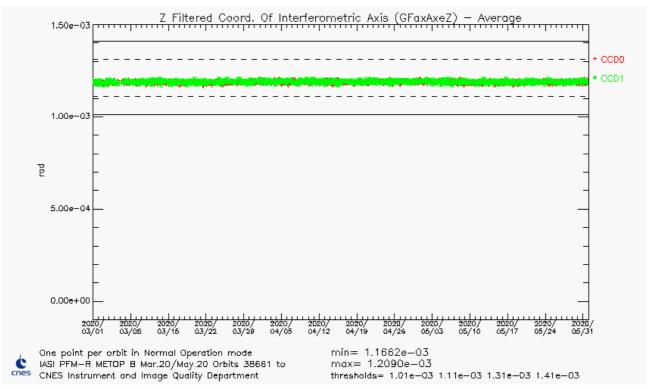
Doc n°: Issue:

Issue: 1.0

Date: 01/06/2020

Sheet: 44 Of: 49

IA-RP-2000-4510-CNE



Metop-B

Figure 33: GFaxAxeZ average (Z filtered coordinates of sounder interferometric axis)

4.6.1.2 Cube Corner constant offset

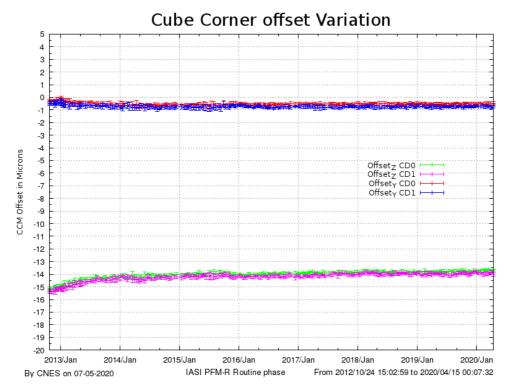


Figure 34: Cube Corner offset variation

Reference cube corner offsets, used in the spectral database of the period (ODB 12), are -0.48 μ m, -0.61 μ m, -14.54 μ m and -14.64 μ m, respectively for Y CD0, Y CD1, Z CD0 and Z CD1. Since the drift of cube corner offset is lower than 4 μ m, there is no need to update the spectral database.





Metop-B

Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

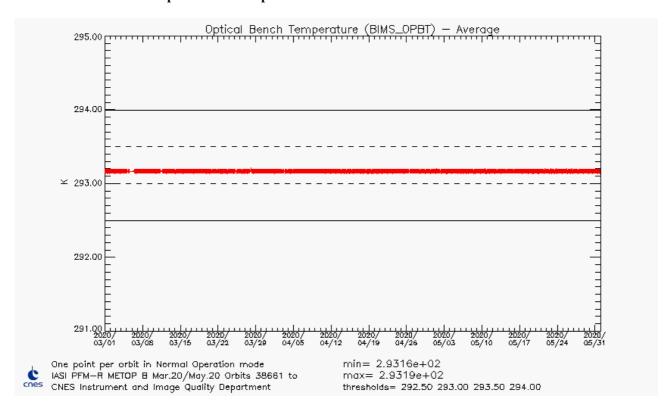
Date: 01/06/2020

Sheet: 45 Of: 49

4.6.1.3 Cube corner velocity

Refer to REVEX, paragraph 5.5.

4.6.1.4 Interferometer optical bench temperature



 $Figure\ 35: Optical\ bench\ Temperature$

4.6.2 **Spectral calibration assessment**

Absolute spectral calibration assessment and Interpixel spectral calibration assessment are performed during routine External Calibration on Earth views at nadir (SP 15) and synthesized once a year.

Refer to REVEX, paragraph 6.6.2.

4.6.3 Conclusion

All parameters impacting IASI spectral calibration are stable and within specifications.





Issue: 1.0

Date: 01/06/2020

Of:

49

Sheet: 46

4.7 GEOMETRIC PERFORMANCES

The geometric calibration is performed on ground (level 1 processing). Most of the analyses of geometric performances require being in external calibration mode.

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Specifications are the following: the IIS/AVHRR co-registration has to be better than 0.3AVHRR pixel while the IIS/sounder co-registration has to be better than 0.8mrad.

4.7.1 Sounder / IIS co-registration monitoring

This monitoring is performed one time a year for REVEX.

The sonder/IIS coregistration error is lower than $100\mu rad$ (eq. 100m on ground).

4.7.2 IIS / AVHRR co-registration

The IIS/AVHRR co-registration is permanently estimated by the L1 processing chain.

Note that AVHRR channels 4 and 5 are within the IIS spectral filter. The spatial resolution of the IIS (0,7km) is close to AVHRR (1km).

The IIS/AVHRR offset guess in the ground segment configuration is used when the algorithm of correlation between IIS and AVHRR does not converge (typically over homogeneous scenes).

The following figures show a comparison of IIS-AVHRR offsets (GlacOffsetIISAvhrr) mean profiles.

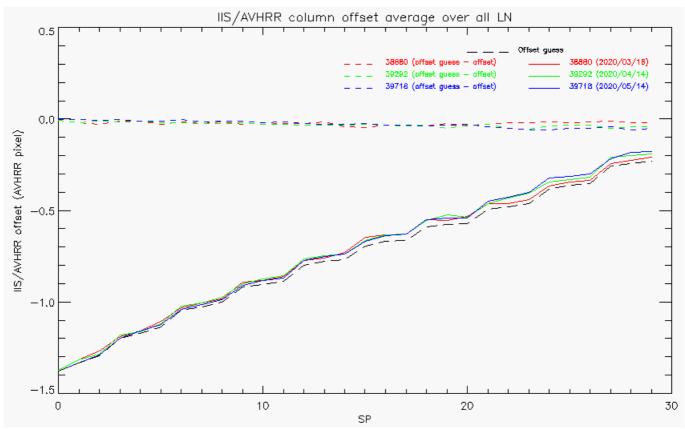


Figure 36 : Column offset (black) guess vs. column offset averaged over all lines (LN) as a function of the scan position (SP=SN), and orbit number (along track IIS/AVHRR offset)





Doc n°: Issue:

sue: 1.0

Date:

Of:

49

IA-RP-2000-4510-CNE

01/06/2020

Metop-B

Sheet: 47

Figure 37: Line offset guess (black) vs. line offset averaged over all lines (LN) as a function of the scan position (SP=SN), and the orbit number (across track IIS/AVHRR offset)

For both across track and along track, the residuals between measured and IIS/AVHRR offset guess in the ground segment configuration are lower than 0.1 AVHRR pixel for all viewing angles, that is equivalent to 100m on ground.

The values are stable.

4.7.3 Conclusion

The positions of IASI pixel are considered stable and well within specification.

IIS-sounder co-registration is stable at about 100μrad which is equivalent to 100m on ground (specification : < 0.8 mrad).

IIS-AVHRR offset is lower than two pixels and stable over time: less than 0.1 AVHRR pixels over three months (specification: < 0.3 AVHRR pixel).

IASI pixel centre location accuracy in AVHRR raster is around 200m. The geolocation of IASI pixels are thus considered stable and well within specification (5 km).





Issue: 1.0

Date: 01/06/2020

Sheet: 48 Of: 49

4.8 IIS RADIOMETRIC PERFORMANCES

The main task of IIS is to insure a good relative positioning of IASI sounder pixels with respect to AVHRR. Its performances are studied on one monthly external calibration in three and the monitoring is performed one time a year for REVEX.

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4.8.1 IIS Radiometric Noise Monitoring

Refer to REVEX, paragraph 6.9.1.

4.8.2 IIS Radiometric Calibration Monitoring

Refer to REVEX, paragraph 6.9.2.

5 IASI TEC SOFTWARE AND INTERFACES

5.1 IASI TEC EVOLUTION

No evolution within the period.

Previous software evolutions:

IASI TEC software version	implementation	Comments
9.4	February 2020	Products browser updates / new ftp logs

Table 19: IASI TEC at CNES Toulouse

5.2 EUMETCAST INTERFACE

EUMETCast dissemination is used for Near Real Time data reception by IASI TEC at CNES, Toulouse. Each orbit, L1 ENG, L1 VER, and AVHRR 1B products are received under continuous series of 3 minutes PDU. Full dumps are reconstructed by the EUMETCAST terminal and pushed to a IASI TEC server. Since August 2012, NPP/CrIS PDU are also received to perform inter-comparison with IASI.

In case of failure of the prime EUMETCAST station, products remain available several days on a redundant station.

The behaviour of the EUMETCAST reception is nominal.

No recent modifications in the EUMETCAST configuration within the period:

Date	EUMETCAST configuration

Table 20: EUMETCAST configuration at CNES Toulouse

5.3 FTP INTERFACE

IASI L0 full dumps are available in Near Real Time on a EUMETSAT FTP server. The IASI TEC software automatically downloads products from the EUMETSAT FTP server.

The reception of L0 products at IASI TEC is nominal.





Metop-B

Doc n°: IA-RP-2000-4510-CNE

Issue: 1.0

Date: 01/06/2020

Sheet: 49 Of: 49

6 CONCLUSION AND OPERATIONS FORESEEN

Please visit https://iasi.cnes.fr to get IASI news.

6.1 SUMMARY

The IASI PFM-R instrument is fully operational.

The instrument configuration is the nominal one.

The main events are:

- SSR anomaly from 27 March 23h39 to 28 March 3h50
- SQ2 on 2 April 2020
- DPS Pixel mode Acquisition anomaly on 27 April 2020
- In Plane manoeuvre on 13 May 2020

6.2 SHORT-TERM EVENTS

- Moon avoidance on 6 September 2020
- Out of Plane manoeuvre on 23 September 2020
- IIS equalization in October 2020

6.3 OPERATIONS FORESEEN

- Update of Scan mirror reflectivity coefficients in 2021
- Next decontamination should happen beginning of 2023

End of document