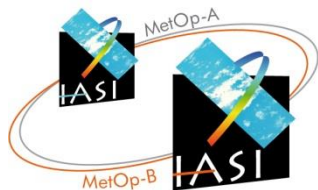


Status of IASI instruments onboard Metop-A and Metop-B satellites



Laurence Buffet⁽¹⁾

Colette Villaret⁽¹⁾

Elsa Jacquette⁽¹⁾

Olivier Vandermarcq⁽¹⁾

Patrick Astruc⁽²⁾

Stephan Anstötz⁽³⁾

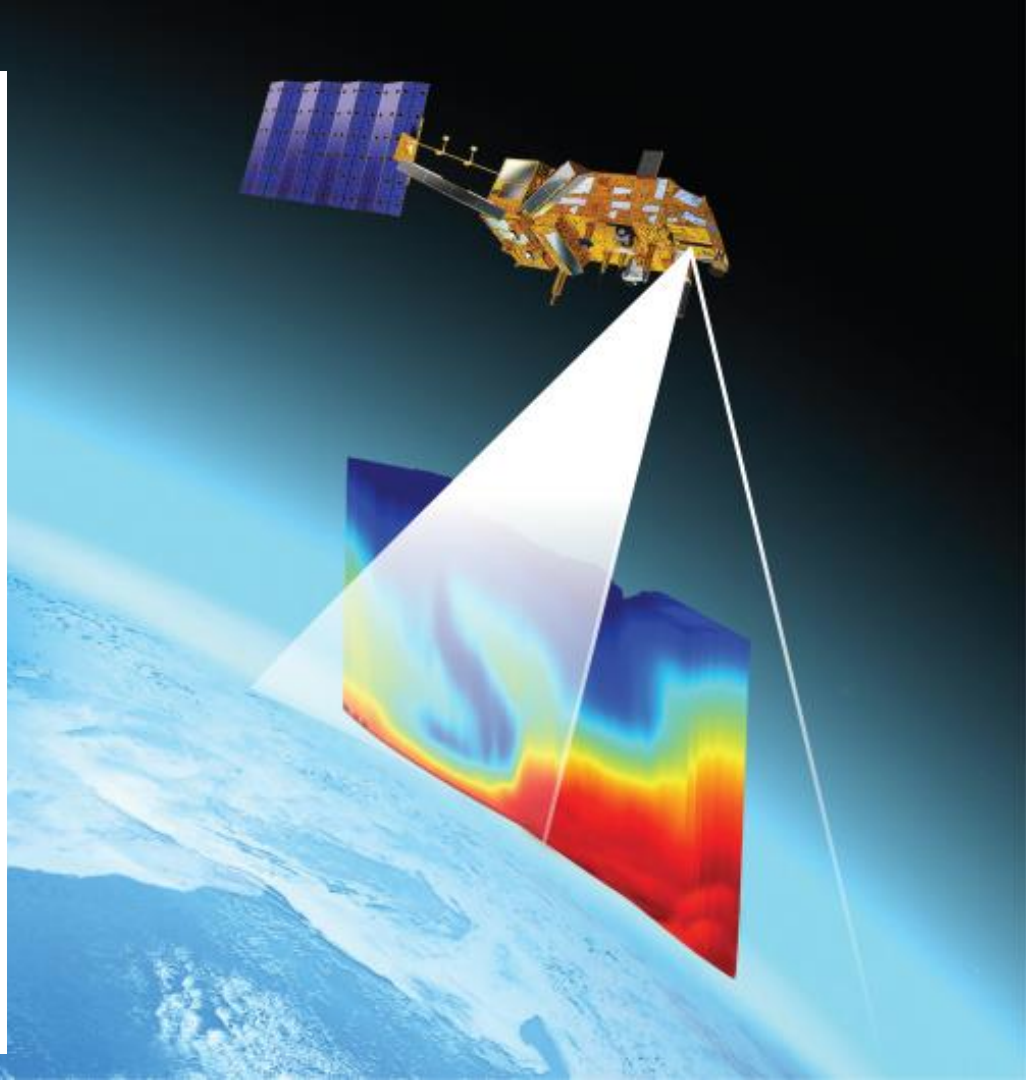
(1)  **cn**es

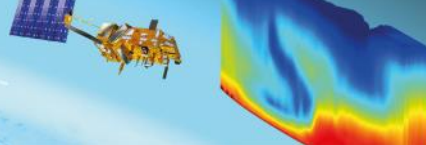
(2)  **ThalesAlenia**
A. Thales / Finmeccanica CartSpace *Space*

(3)  **EUMETSAT**

IASI 2016

11-15 April 2016 Antibes Juan-les-Pins





2 operational instruments

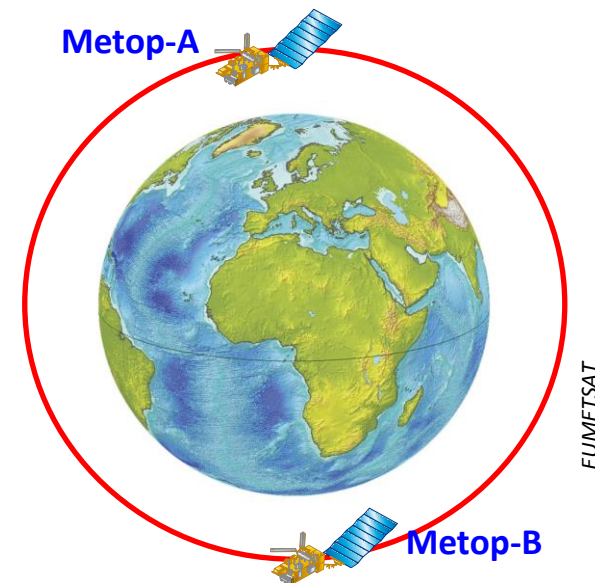
- **IASI-A since July 2007** (launched October 2006)
- **IASI-B since April 2013** (launched September 2012)

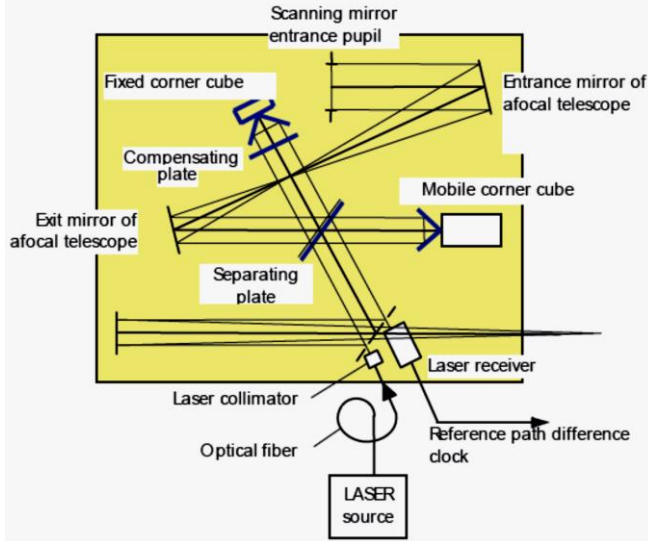
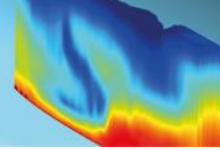
Mission extended for IASI-A : specified lifetime was 5 years

Both instruments deliver **very good performances**, **similar** between each other and with a **very good stability in time**. We also aim at maintaining a maximal operational **availability** for both instruments.

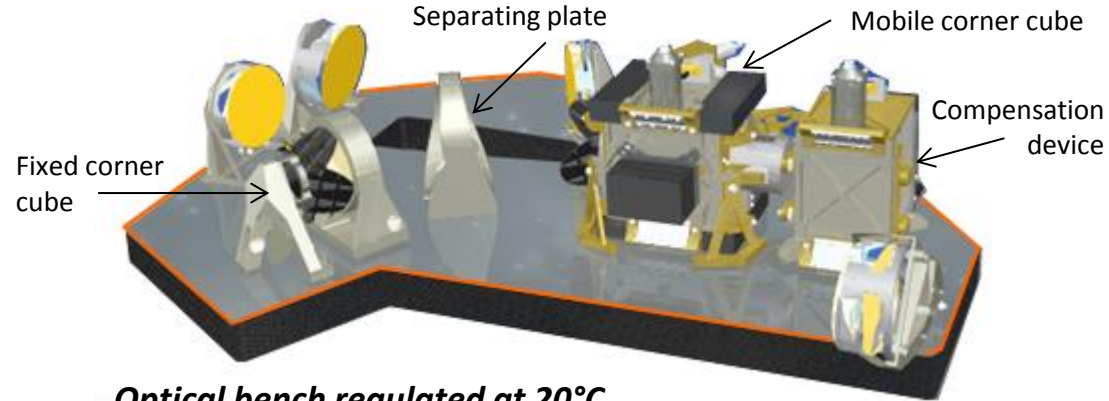
➔ **Performance status : poster S1-03 (Claire Maraldi)**

A **constant quality of distributed data** is guaranteed via an accurate monitoring of instrument performance indicators done by IASI TEC at CNES on L1 products. In addition, all the instrument housekeeping telemetries are ground monitored by EUMETSAT.



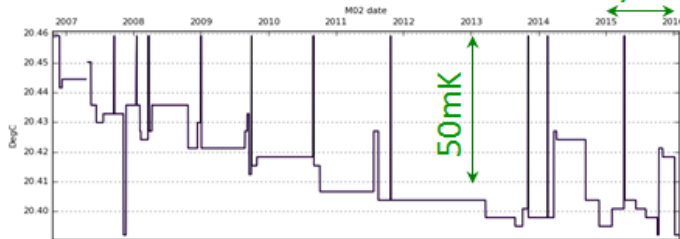


Interferometer stability (1/2)

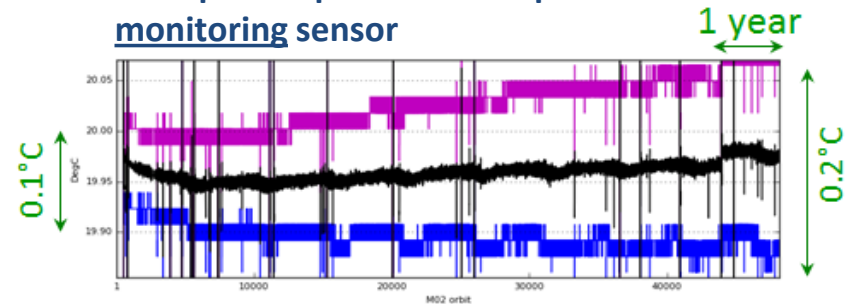


Optical bench regulated at 20°C

Reference laser : temperature loop stability 1 year

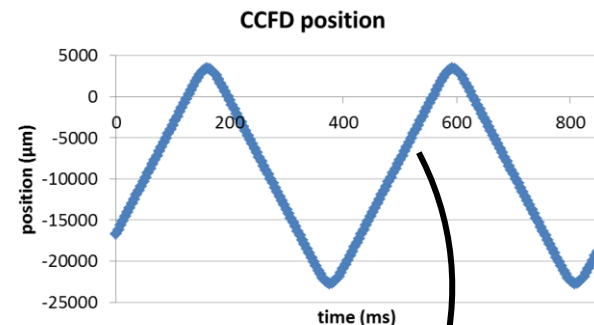
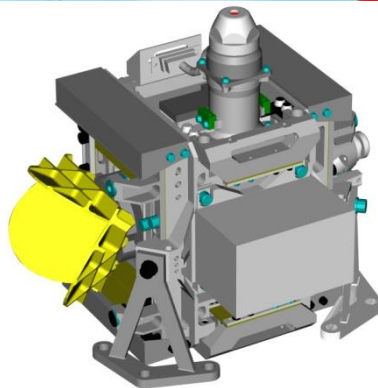


Example of optics area temperature monitoring sensor 1 year

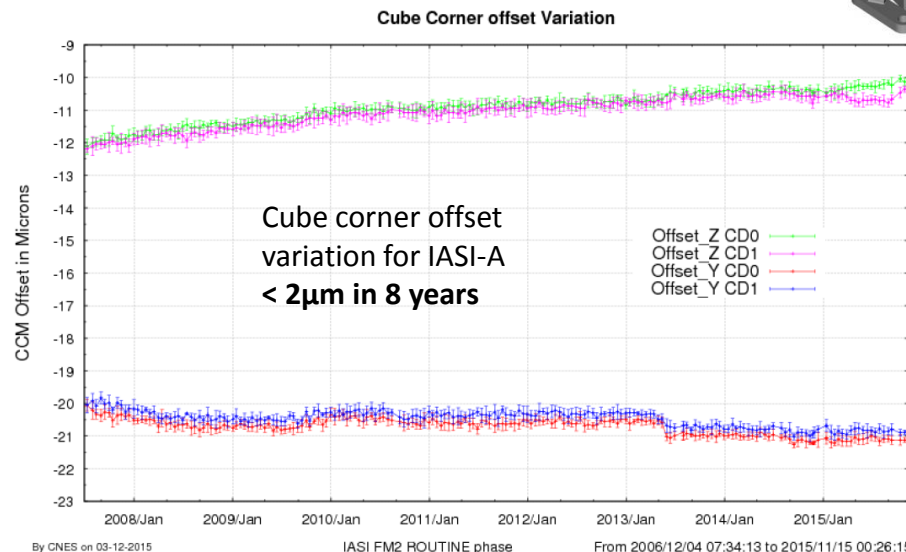
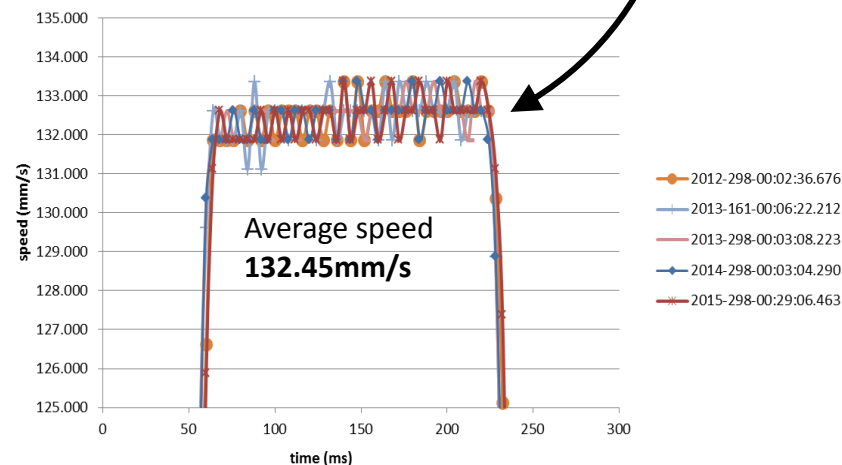


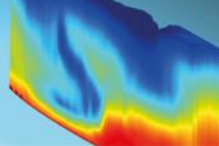
Interferometer stability (2/2)

- **Stability of Corner cube mechanical performance**
 - Stable speed profile in both directions
- **Stability of Corner cube alignment**
 - Very low drift for both IASIs



IASI PFMR/Metop-B flight data
CCFD speed - Forward direction

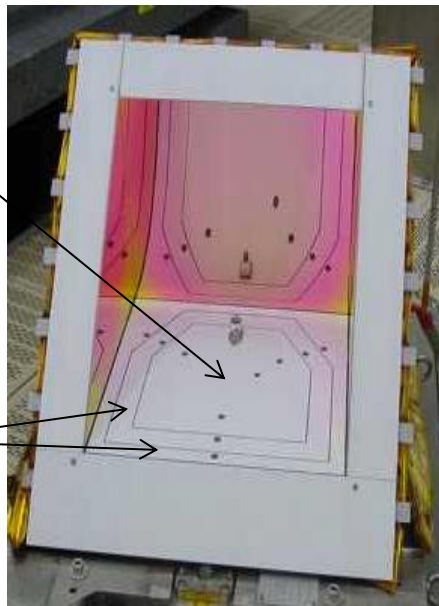




CBS : 3 stages passive cryo-cooler

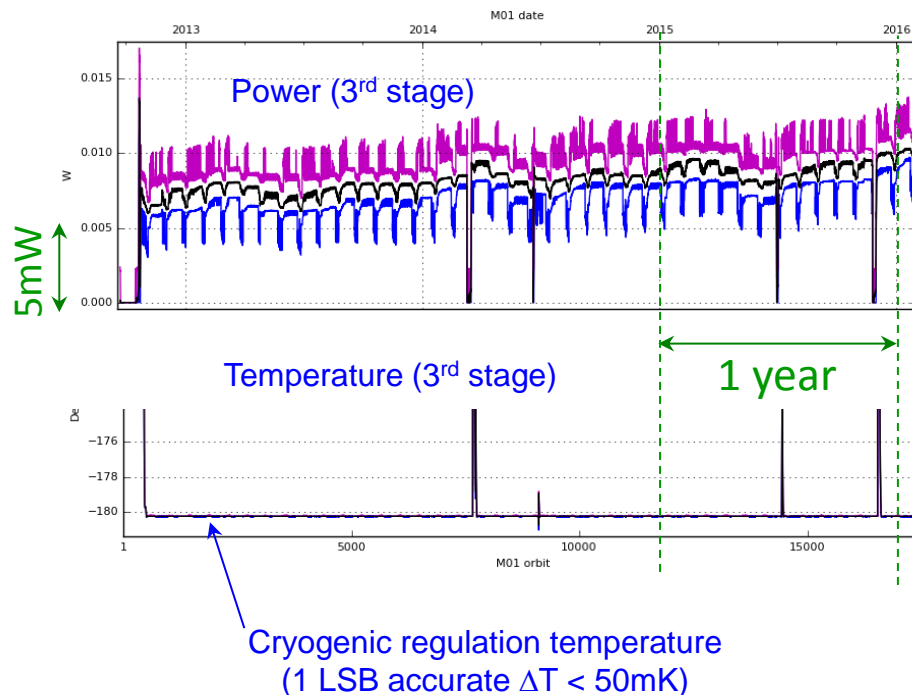
3rd stage regulated
temperature
(=detector location
stage) : **perfectly
stable at the target
(close to 90K)**

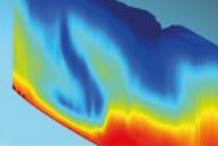
2nd and 1st stages
unregulated
temperature
(decoupling stages) :
**stable (evolution
<+0.2°C per year)**



Detectors temperature stability

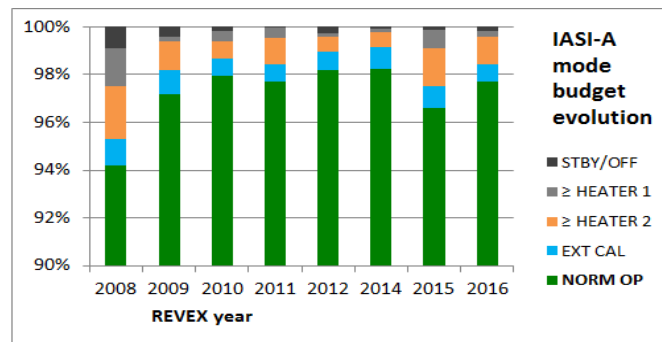
Example of IASI-B





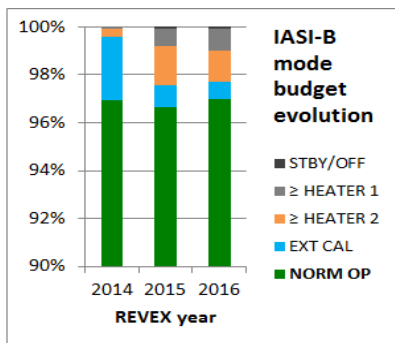
Instrument availability

Instrument in **Normal Operation mode**
≈ 97% for each IASI



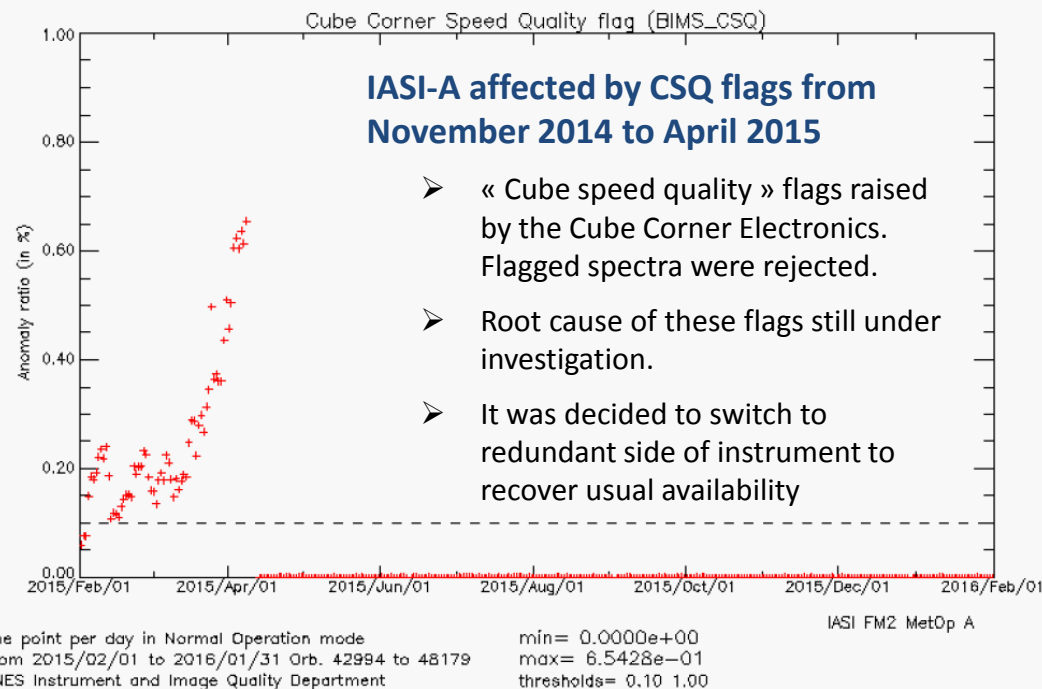
Main outage contributors

- Decontamination
- Anomalies
- Routine calibration
- External events



and

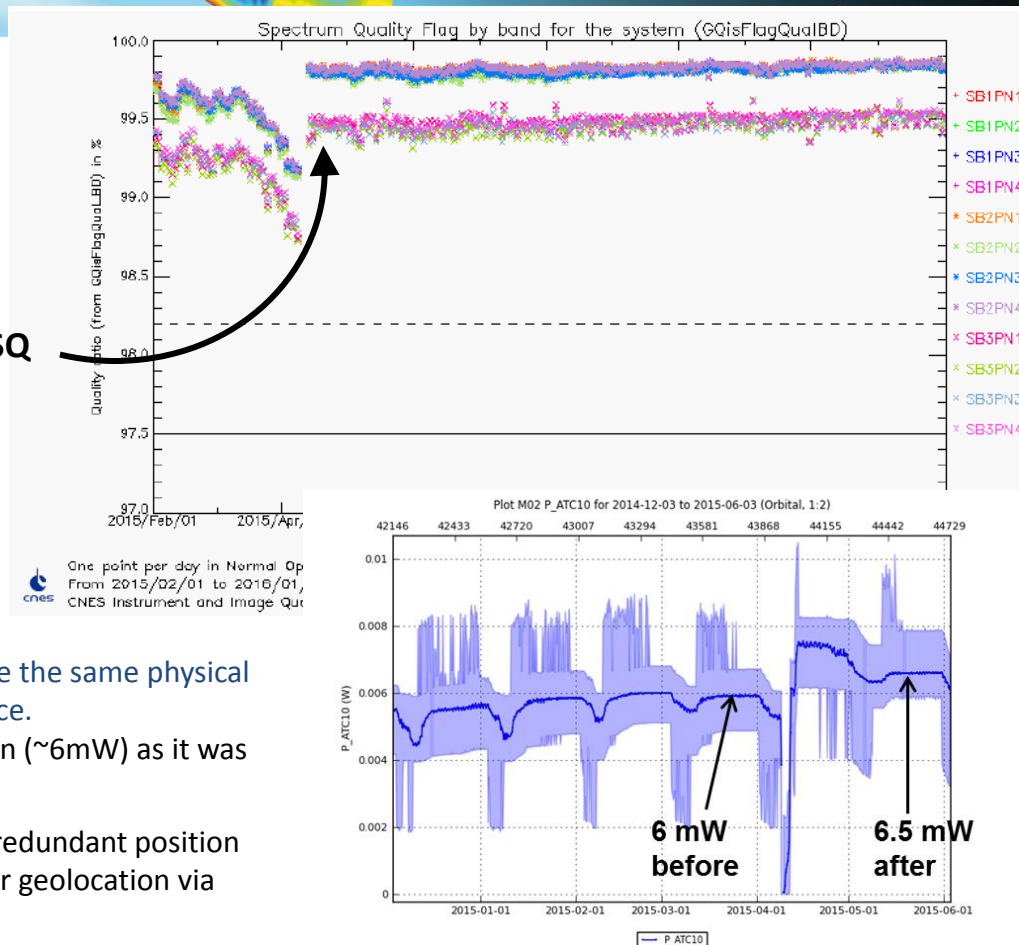
Quality of L0 spectra : usually $\leq 0.5\%$ rejected spectra (mainly due to spikes in B3)

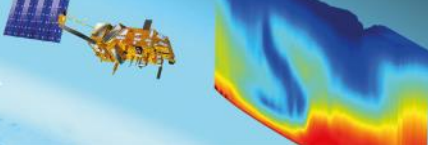


Main changes onboard since 3rd IASI conference (2013)

IASI-A switched to redundant side (April 2015)

- ➔ Availability much improved : **almost no more CSQ flags**, availability restored to its maximum.
- First time IASI-A was operated on redundant side
 - Redundant thermal control
 - Redundant electronics
- **Performance identical on redundant side**
 - Detectors regulation target adjusted in order to have the same physical temperature to maintain B1 radiometric performance. Regulation power adjusted close to minimum margin ($\sim 6\text{mW}$) as it was on nominal side.
 - Small effect on IASI Line Of Sight because of use of redundant position encoder for scan mechanism : taken into account for geolocation via IIS/AVHRR offset guess update



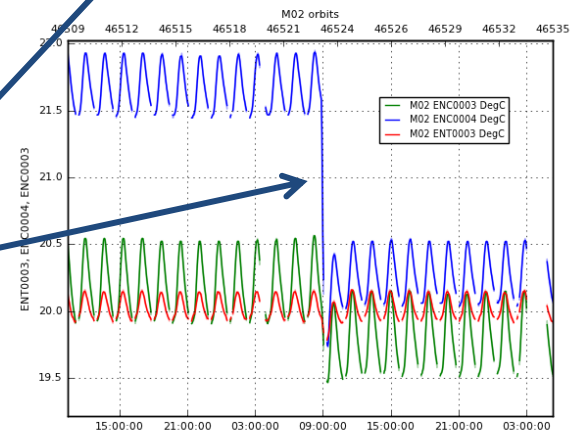
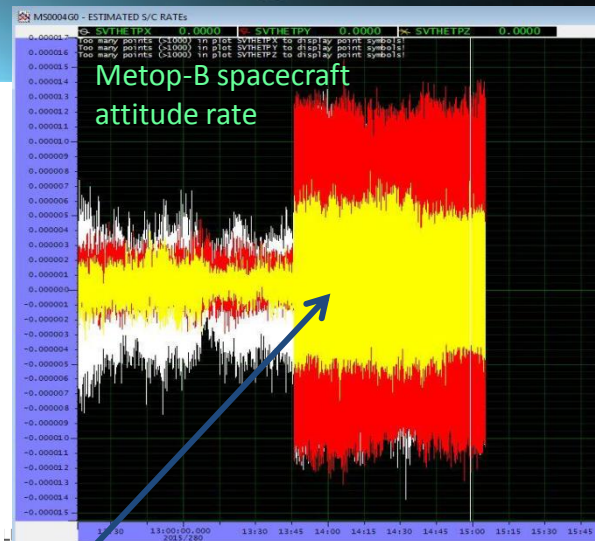


Main changes onboard since 3rd IASI conference (2013)

Compensation Device stopped permanently for both IASI's since October 2015

➔ Effect on IASI performance: poster S1-08 (Elsa Jacquette)

- Compensation Device mechanism was designed to compensate for Cube Corner mechanism exported efforts towards the satellite, in a configuration with LFD released (which is not the case for both IASI's)
- CD stop was not an option in original IASI design. A bypass solution was found to make the CD stop.
- **CD stop configuration was tested first on IASI-A in November 2014 and found to have a positive impact on IASI performance** (suppression of « ghost » effect due to separating blade microvibration in the interferometer), and an **acceptable effect on Metop satellite AOCS**. Thus a **permanent stop** was decided and **performed in October 2015**.
- Visible via instrument telemetries
 - small local decrease of CD temperature
 - no side effect on CCFD and optical bench thanks to thermal regulation

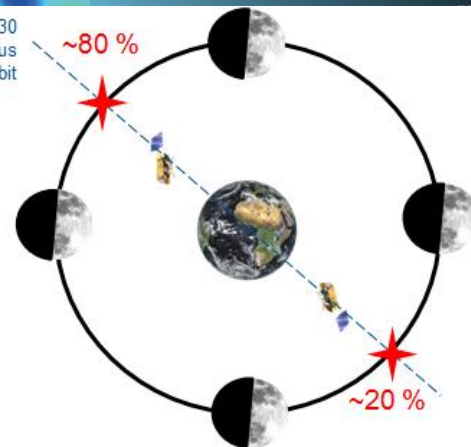


Some proposals for IASI-A End of Life test preparation

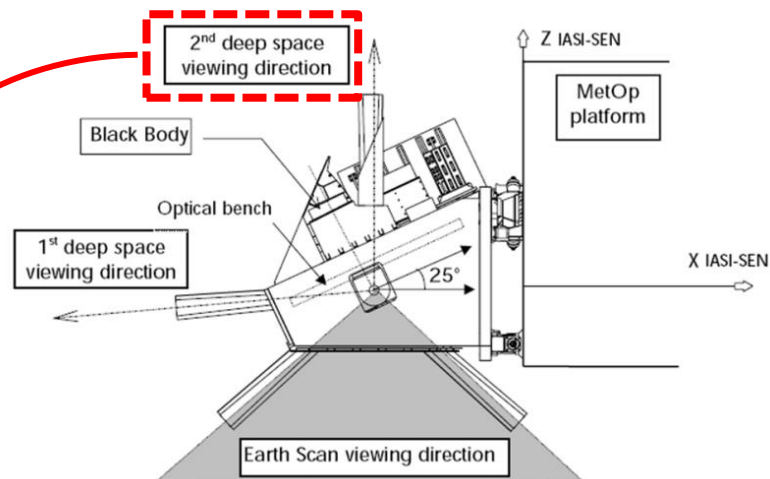
Acquisition of Moon in Cold Space view (to be studied)

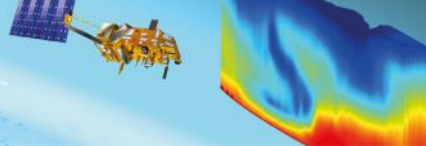
- Possibility to test **new calibration methods for TIR** already used in VIS and SWIR.
- Could be extended to **inter-compare IASI-A and IASI-B** calibration using the Moon (relative calibration).
- Derive IR spectra of the Moon and thus **scientific information about the Moon** (surface temperature and composition). Proposal to characterize Moon in TIR domain is supported by GSICS.

Plan of 21h30
sunsynchronous
orbit



Example of Moon transit in IASI pixels (taken from IASI-B Cal/Val)





Some proposals for IASI-A End of Life test preparation

Local improvement of spatial sampling and possibly spatial resolution (to be studied)

- **Reduction of swath and increase of spatial sampling** by scan parameter modification.
- Final objective: **possibility of improvement of spatial resolution by ground processing of overlapping acquisitions** ("supermode" like)

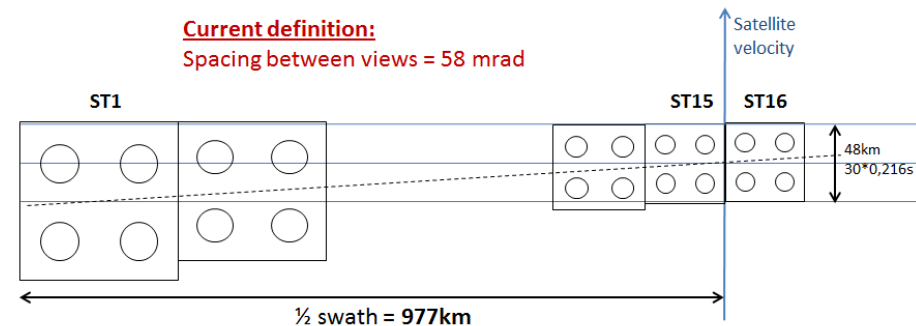
- Concept to be analysed in detail

- Need of sufficient accuracy of pointing and geolocation
- Need of sufficient knowledge of instrument IPSF

- A first study could be done on available data from nadir acquisitions during monthly External Calibration (oversampling along satellite velocity)

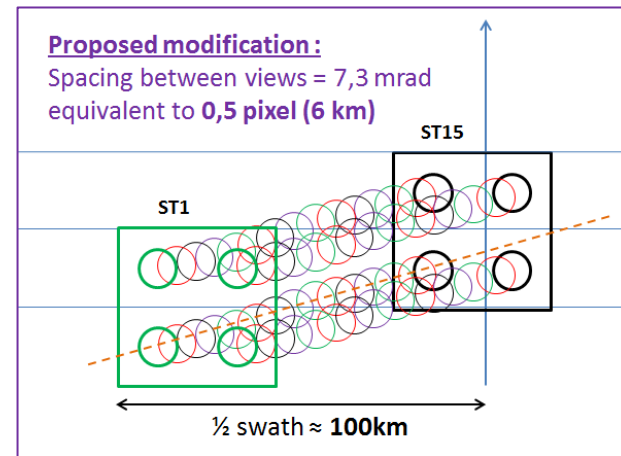
Current definition:

Spacing between views = 58 mrad

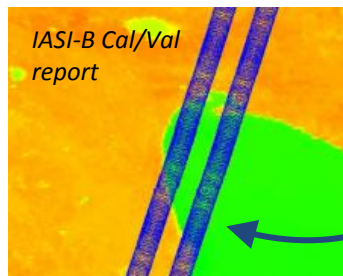


Proposed modification:

Spacing between views = 7,3 mrad
equivalent to 0,5 pixel (6 km)



IASI-B Cal/Val
report





Credits ThalesAleniaSpace

Third IASI model to be delivered

- **IASI FM3 retrofit** : replacement of old NdFeB magnets inside scan and corner cube mechanisms, by new magnets less sensitive to corrosion
- Delicate intervention on corner cube mechanism implied **interferometer realignment**: successfully performed
- Instrument acceptance campaign achieved at TAS. Instrument final performance validated by **Optical Vacuum Test in January 2016**.
 - **Very good radiometric and spectral results**, not affected by the retrofit operation



➔ **IASI FM3R will be ready for reintegration on Metop-C PLM in September 2016, followed by satellite AIT campaign then launch in October 2018**